

Mark Tuttle Dept. Mechanical Engineering University of Washington

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<u>Outline</u>

- I. Brief summary of a UW numerical study funded by Boeing in 2003
- II. Experimental studies at UW, 2008-09
- III. Newly-funded JAMS-AMTAS project, initiated September 2015



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<u>Created a finite-difference program called MOIST; predictions</u> <u>based on</u>:

- Fick's diffusion equations → used to predict through thickness moisture content resulting from cyclic changes in external temperature and humidity
- Fourier heat conduction equation → used to predict steadystate through-thickness temperature profiles
- Clapeyron equations → used to predict dew point of water vapor in core (if current temperature < dew point, condensation occurs)
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Overall Conclusions:

- Core humidity levels will increase with time...only question is time required
- Under realistic service conditions liquid water does not accumulate within core region of pristine structures due strictly to diffusion...<u>however</u>
- (For transport aircraft flight profiles) humidity increase implies water vapor will condense-freeze-thaw-evaporate during ascentcruise-descent cycles

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Predicting Moisture **S** Diffusion

Advanced Materials in Transport Aircraft Structures and Humidity



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AS Predicting Moisture Diffusion

Typical Result: Constant

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Tomporatures and Humidity (cont'd)





Although reasonable predictions were obtained using program MOIST, no experimental measurements were available to validate analysis

The potential structural application of sandwich composites envisioned by Boeing engineers (in 2003) were not pursued, funding was discontinued, and the UW study ended.



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Objectives:

Measure relative humidity in core region of a flat sandwich panel exposed to constant external temperature and humidity on both sides

Compare measurements with MOIST predictions

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Sandwich panel internally instrumented with:

- 2 type K thermocouples
- 2 Ohmic Instruments Model HC-610 capacitive humidity sensors:
 - 5-95 %RH
 - -40 to 185°F operating range

Test begun on 5 Aug 2008:

- Temperature set @ 40°C (104°F)
- Humidity level 55%RH
- Data recorded every 30 mins (using Labview)

HC-610 Thermoset polymer capacitive humidity sensor. Hybrid electronics. Linear output. Range 5 to 95 %RH 2%. Temp. – 40 to 185 °F. Supply voltage 4.0 - 5.8 VDC



PDF Man/Instructions



www.ohmicinstruments.com/







- Type 410 Nomex honeycomb core
- [0/45/90/-45]_s Gr/Ep facesheets
- Core sized to fit within aluminum frame to insure 1-D, through-thickness diffusion
- First facesheet bonded to one side of panel using thin-film adhesive
- Pocket for embedded humidity sensors and thermocouples milled in core





 Leadwires inserted through honeycomb and aluminum frame



 Initial installation of embedded sensors





• Leadwire passage in aluminum frame sealed with epoxy



 Honeycomb 'caps' placed over instrumented sites





 Second facesheet bonded to panel using thin-film adhesive...





...and hot press

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- Completed panel mounted in test chamber; exposure began on 5 Aug 2008
- Initial measurements (40°C = 104°F): Panel hum sensor 1: 25.0 %RH Panel hum sensor 2: 23.4 %RH (...higher than anticipated in '03...)





S Typical Measurements Humidity data from 7:50am 3 Nov to

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Advanced Materials in Transport Aircraft Structures (48 hr period, about 3 months after test

initiated)





Typical Measurements

Humidity data from 7:50am 3 Nov to 7:50am 5 Nov

(48 hr period, about 3 months after test

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Aug'09 = 365 days





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- Change in bending stiffness, EI_{eff} (measure using 4-pt bend test)
- Change in G_{I} (measure using methods being developed by CMH-17 group)



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• Produce 32, 2 in x 12 in specimens (16 autoclave-cured, 16 oven-cured with vacuum bag)

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Schematic of experimental arrangement to measure G_I for sandwich panels (under development by CMH-17 working group)





ps of test setup at NIAR





Effects of Moisture Diffusion in Sandwich Advanced Materials in Transport Aircraft Structures Composites

Photos of test setup at NIAR





Thank You!

Comments/Questions/Suggestion s?



Backup Slides



Effects of Moisture **Diffusion** in Sandwich Advanced Materials in Transport Aircraft Structures Composites



American Autoclave featuring 42 in dia x 96 in working chamber

Effects of Moisture **TAS** Diffusion in Sandwich Advanced Materials in Advanced Materials in Transport Aircraft Structures Composites



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Blue-M Model POM-246F Lab Oven



Wabash Model G50H-24-BCLX 50-ton hot press





Cincinnati Sub-Zero "Tundra" Environmental Conditioning Chamber



Thermotron Model S.12 Temperature Chamber





Instron Model 8511 hydraulic test frame

Advanced Materials in Transport Aircraft Structures Study Funded by Boeing in 2003

<u>Assumptions</u>:

- 'Pristine' (undamaged) sandwich structure; moisture ingression solely due to diffusion
- Core = Nomex honeycomb
- Initial moisture content = 0% . This implies:
 - Initial moisture content of composite face sheets = 0%
 - Initial relative humidity within core region = 0%



Typical Predicted AS Steady-State **Temperature Profiles**

12-ply Gr-Ep facesheets (0.060 in thick)

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- Nomex honeycomb core (0.50 in thick)
- <u>Step 1:</u> Inside temp = 70FOutside temp = 85F

<u>Step 2:</u> Inside temp = 65FOutside temp = -65F



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Example:

RH = 40%

Temp = 70 F

Example:

RH = 80%

Temp = 90 F

Example:

Temp = 90 F

RH = 80%

Example:

RH = 0%

Temp = -65 F

Advanced Materials in Transport Aircraft Structures Temperature and Humidity

• <u>Step 1:</u>

Duration: 120 minutes Inside: temp = 90F; RH = 80% Outside: temp = 90F; RH = 80%

- <u>Step 2:</u> Duration: 380 minutes
 Inside: temp = 70F; RH = 40%
 Outside: temp = -65F; RH = 0%
- (Total cycle time = 500 minutes = 8.3 hrs)
- 12-ply Gr-Ep facesheets (0.060 in thick)
- Nomex honeycomb core (0.50 in thick)
- Initial moisture content assumed = 0%







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 Through-thickness temperature distribution assumed to be governed by the Fourier heat conduction equation:

 $\partial Q / \partial t$ = heat transfer rate

 K_z = thermal conductivity (z - direction) $A = a_{b}^{\partial Q} e^{-K_z A} \frac{\partial T}{\partial z}$

where:

$$T = temperature$$

t = time

 Material properties allowed to vary through thickness; heat conduction equation solved numerically using finite-differences



- *K*_{core} estimated using rule-of-mixture approach
- Volume fractions of air and paper within the core calculated using hexagonal repeat unit





Repeat Unit

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 Given the cell size (c), paper ribbon thickness (w), and core thickness (t), it can be shown that the volume fractions are given by:



 $V_{air} = \frac{(3c - 8w)}{3c}$ $V_{paper} = \frac{8w}{3c}$

Advanced Materials in Transport Aircraft Structures Conductivity

Example: $K_{core} = (V_{air})(K_{air}) + (V_{paper})(K_{paper})$ Honeycomb core with 0.20 in cell size, produced using 2-mil thick DuPont Type 410 Nomex paper:

$$K_{paper} = 0.715 \text{ BTU-in/hr-ft}^2-^{\circ}\text{R}$$

 $K_{air} = 0.166 \text{ BTU-in/hr-ft}^2-^{\circ}\text{R}$

Calculated quantities:

$$V_{paper} = 0.027$$

 $V_{air} = 0.973$

$K_{core} = 0.181 \text{ BTU-in/hr-ft}^2-^{\circ}\text{R}$

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* Nomex properties: http://www.matweb.com Air properties: *Marks' Standard Handbook for Mechanical Engineers*, 8th Ed (1978)



Material	Thermal Conductivity, K (BTU-in/hr-ft ² -°R)	Thickness (in)
Graphite/Epoxy	4.0*	0.005 (ply)
Honeycomb Core	0.181	0.50

* <u>Note</u>: Typical through-thickness K for Gr/Ep is listed; in-plane K values typically > 400 BTU-in/hr-ft2-°R

Advanced Materials in Transport Aircraft Structures Predicting Moisture Diffusion

 Through-thickness (1-D) diffusion of moisture assumed to be governed by Fick's first and second laws:

$$\phi = D_z \frac{\partial c}{\partial z} \qquad \qquad \frac{\partial c}{\partial t} = \frac{\partial}{\partial z} \left[D_z \frac{\partial c}{\partial z} \right]$$

 ϕ = rate of diffusion ("moisture flux") : units = mass/(area * time)

- c = concentration : units = (mass/volume)
- $D_z = diffusivity : units = area/time$
- z = direction of diffusion : unit = length

t = time

Advanced Materials in Transport Aircraft Structures Advanced Materials in Diffusion

 From an experimental point of view it is easier to deal with percent moisture by weight (M), rather than the concentration of moisture (c).
 Fick's first and second laws are restated as:

$$\phi = \frac{D_z \rho}{100} \frac{\partial M}{\partial z} \qquad \frac{\partial M}{\partial t} = D_z \frac{\partial^2 M}{\partial z^2}$$

 ρ = density, mass/volume

M = "moisture content"

$$M = \frac{(\text{current weight}) - (\text{dry weight})}{(\text{dry weight})} \times 100\%$$



Temperature dependency of diffusion coefficient for solids (i.e., ply and core paper) assumed to follow a Arrhenius-type relationship:

$$D = D_o \exp\left(-\frac{E}{T}\right)$$

where: D_o , E = known material constants (differ for ply and core paper) 53

T = absolute temperature

Advanced Materials in Transport Aircraft Structures Advanced Materials in Diffusion

Temperature dependency of diffusion of H₂0 vapor in air assumed to follow a power law of the form*:

$$D_{air} = 0.03376 \left(\frac{T(^{\circ}R)}{491.67(^{\circ}R)}\right)^{1.81} \frac{in^2}{\sec}$$

* Massman, W.J., *Atmospheric Environment*, Vol 32 (6), pp 1111-1127 (1998).



$$\rho_{core} = (V_{air})(\rho_{air}) + (V_{paper})(\rho_{paper})$$

$D_{core} = (V_{air})(D_{air}) + (V_{paper})(D_{paper})$

Advanced Materials in Transport Aircraft Structures Predicting Moisture Diffusion

The moisture content (*M*) of any surface layer in contact with air can be related to the relative humidity according to (Springer, 1980):

$$M = M_u \left(\frac{\% RH}{100}\right)^{\nu}$$

- constant M_u = material property
- exponent b = 1 for most materials
- relationship used to define the boundary condition at all ply interfaces



Predicting Moisture Diffusion

Preceding relations allows forward-difference solution to Fick's equations; summary

- (At all interior ply interfaces) moisture flux leaving ply k must equal moisture flux entering ply k+1 $M = M_u \left(\frac{\% RH}{100}\right)$
- (Boundary conditions):
- (Initial conditions): Initial through-thickness moisture content assumed uniform (assumed = zero in `03)
- Time step increment of 1 minute

Predicting Moisture Diffusion

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Properties Used in '03

Property	Gr/Ep (typical values)	Type 410, 2-mil Nomex (www.matweb.com)
D _o	0.010 in²/sec (see note)	0.006 <i>in²/sec</i>
E	10300 °R	9000 °R
M _u	0.02	0.03
Density, $ ho$	0.054 <i>lbm/in</i> ³	0.026 <i>lbm/in</i> ³

<u>Note</u>: Properties reported for Gr/Ep vary widely. For example: $0.005 < D_o < 0.040 \text{ in}^2/\text{seC}$



within core following Step 1, then

1. Calculate relative humidity within core volume (based on rule of mixtures approximation)

$$\% RH = M_{core} / M_u^{core} = \left(\frac{M_{core}}{V_{air} + M_u^{pap} V_{pap}}\right)$$



2. Use Clapeyron equation to estimate saturated vapor pressure (P_{svp}) at Step 1 core temperatures

$$T_{cor}^{s1} \qquad P_{svp} \approx P_{svp}^{ref} \exp\left[\frac{h_{fg}(T_{cor}^{s1} - T^{ref})}{(RT^{ref})T_{cor}^{s1}}\right]$$

 T^{ref} = ref temperature (e.g., T^{ref} = 70°F = 529.67°R) P_{svp}^{ref} = saturated vapor pressure at T^{ref}

 $= 0.3632 \text{ psi at } T^{ref} = 529.67^{\circ} R$

 h_{fg} = enthalpy of vaporization at T^{ref}

= 1054 BTU/lbm = 820E3 ft - lbf / lbm at $T^{ref} = 529.67^{\circ}R$

R = gas constant for water vapor = 85.76 ft - lbf / lbm - R



Condensation (cont'd)

3. Calculate partial pressure of water vapor at Step 1 temperatures:

$$P_{pp} = (\% RH)(P_{svp})$$



4. Calculate dew point temperature based on reference temperature and partial pressure at step 1 temperatures:

$$T_{dewpt} = \frac{h_{fg}T^{ref}}{h_{fg} - \ln\left(\frac{P_{pp}}{P_{svp}^{ref}}\right)RT^{ref}}$$



5. Finally, condensation is predicted if core temperatures during step 2 become lower than calculated dew point temperature

$$T_{cor}^{s2} < T_{dewpt} \Rightarrow$$
 Condensation