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# Effects of Moisture Diffusion in Sandwich Composites

2016 Technical Review

William Smoot, Sung Lin 'Jason' Tien,

Shuyu 'Frank' Xia, and Mark Tuttle

Department of Mechanical Engineering

University of Washington

# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

- In-service bond failures between composite facesheets and honeycomb cores have been reported in the space, marine, and aviation industries

X-33 Liquid Hydrogen  
Tank Failure

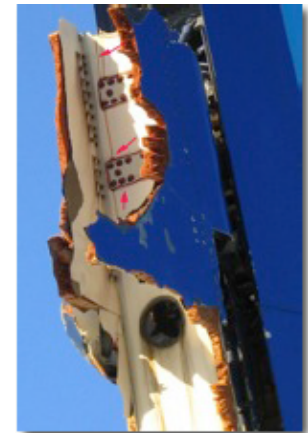


Boeing 747 upper  
skin disbonds



approx. 24" x 60"  
upper skin disbond

Airbus A-310  
Rudder Failure



(Photos courtesy of Ronald Krueger, National Institute of Aerospace)

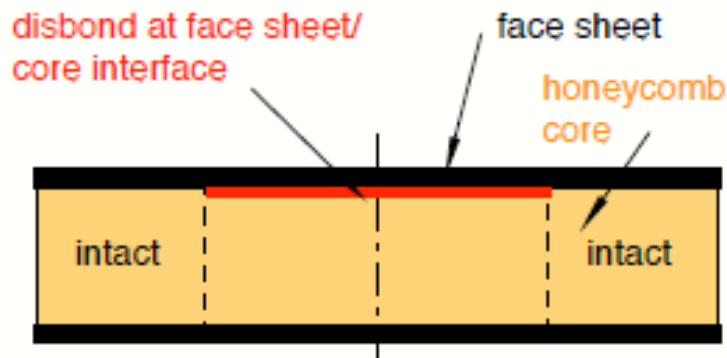
# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

- Core-to-skin disbond initiation and growth are not completely understood, but are thought to occur due to combination of factors:
  - Pressure differences between inside and outside of unvented honeycomb structures (Ground-Air-Ground or 'GAG' pressure cycles)

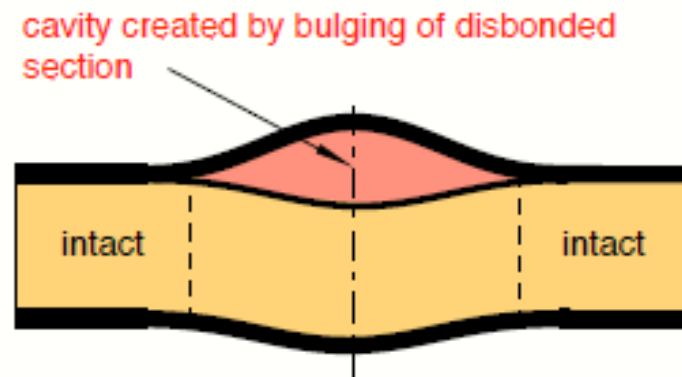
### Configuration at ground level

$$P_o = 100 \text{ kPa} = 14.7 \text{ psi}$$



### Configuration at 35,000 ft

$$P_o = 24 \text{ kPa} = 3.5 \text{ psi}$$



# Effects of Moisture Diffusion in Sandwich Composites

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- Core-to-skin disbond initiation and growth are not completely understood, but are thought to occur due to combination of factors:
  - Pressure differences between inside and outside of unvented honeycomb structures (Ground-Air-Ground or 'GAG' pressure cycles)
  - In-plane (design) loads
  - Water ingress into core, followed by freeze-thaw cycles
- Water ingress most commonly attributed to wicking of liquidous water through microcracks, along fiber/matrix interface, and/or through improper edge closeouts (all accentuated by GAG pressure cycles)
- Water ingress may also occur due to diffusion of water molecules through (undamaged) facesheets



# Effects of Moisture Diffusion in Sandwich Composites

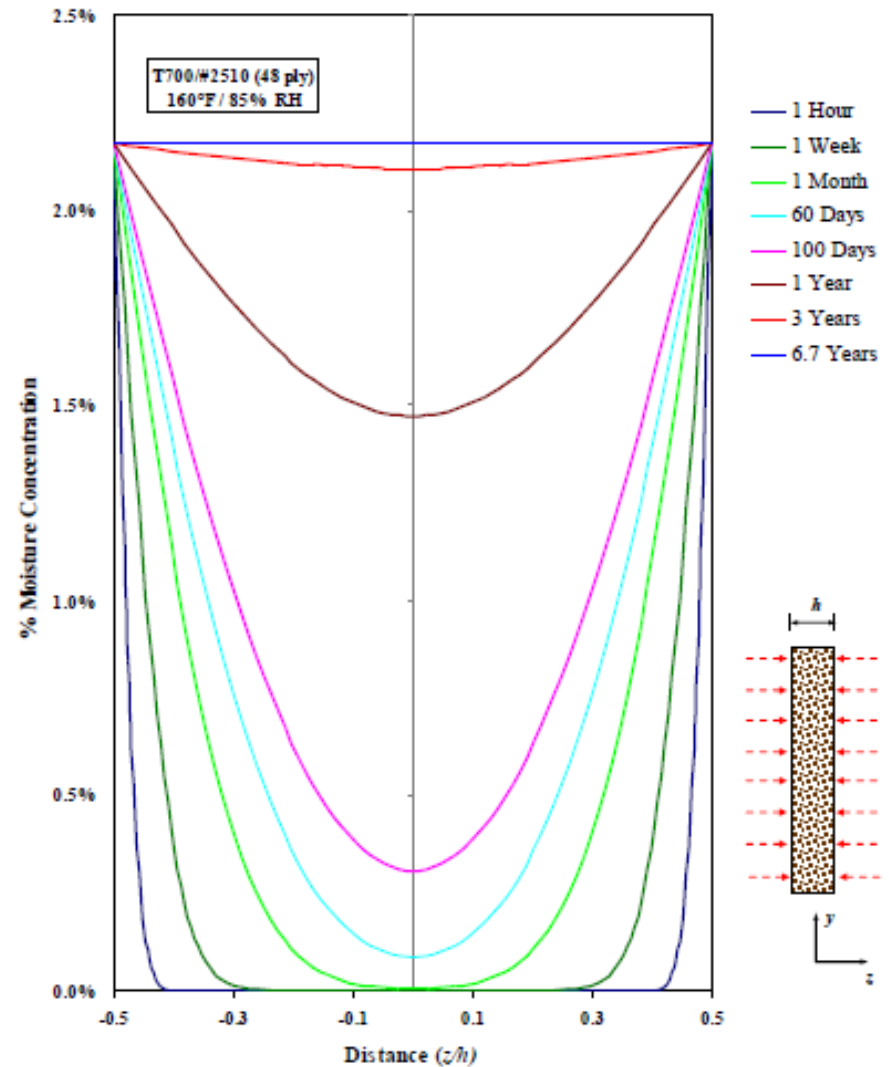
## Motivation and Key Issues:

- Significant moisture transport via diffusion typically requires months or years, depending on:
  - Temperature
  - Thickness and material properties
  - External humidity level

# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

Moisture diffusion in solid 48-ply Gr-Ep laminate; 160°F, 85%RH (W. Seneviratne and J. Tomblin, JAMS 2012)



# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

Moisture diffusion in  
honeycomb

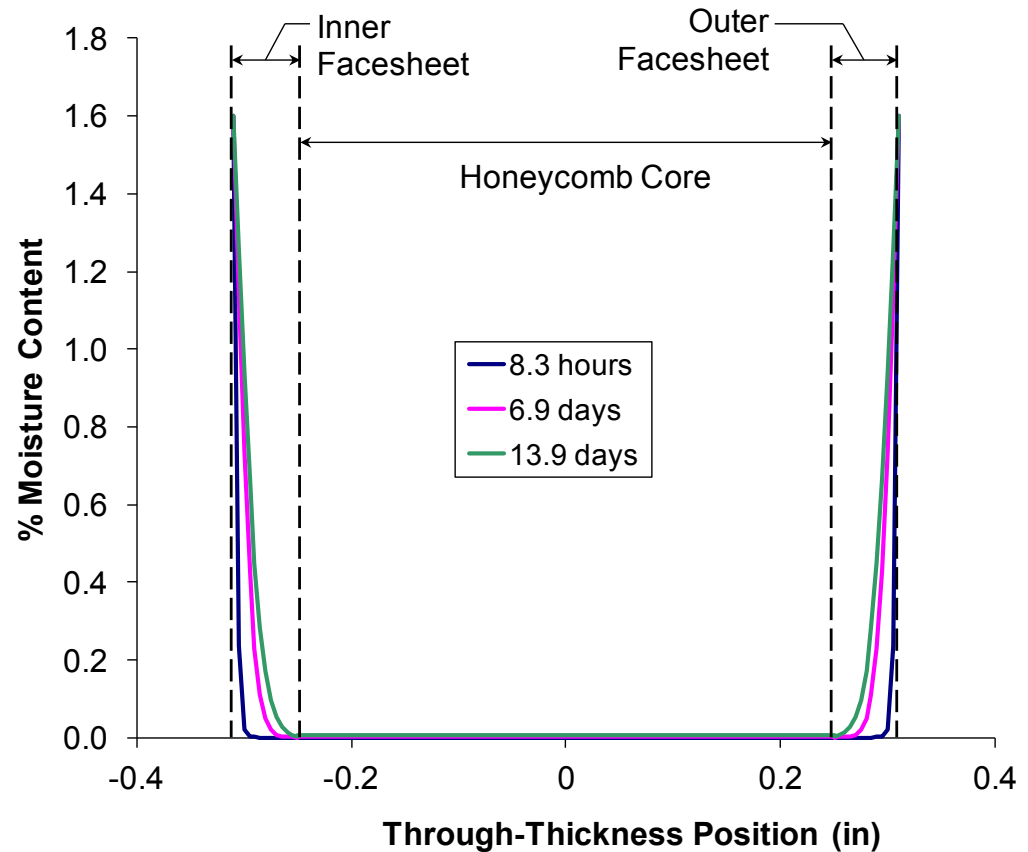
sandwich panel:

- 12-ply Gr-Ep facesheets

- 0.5 in Nomex core

- 90°F, 80%RH

(Tuttle, AMTAS 2009)



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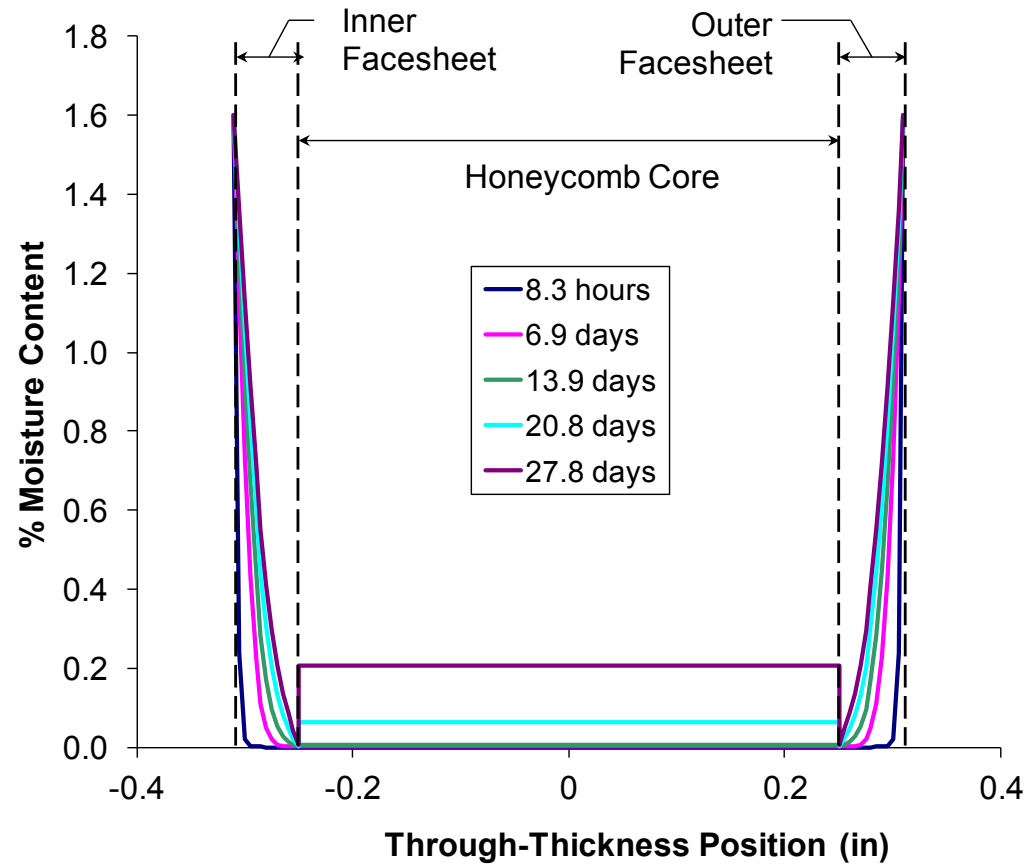
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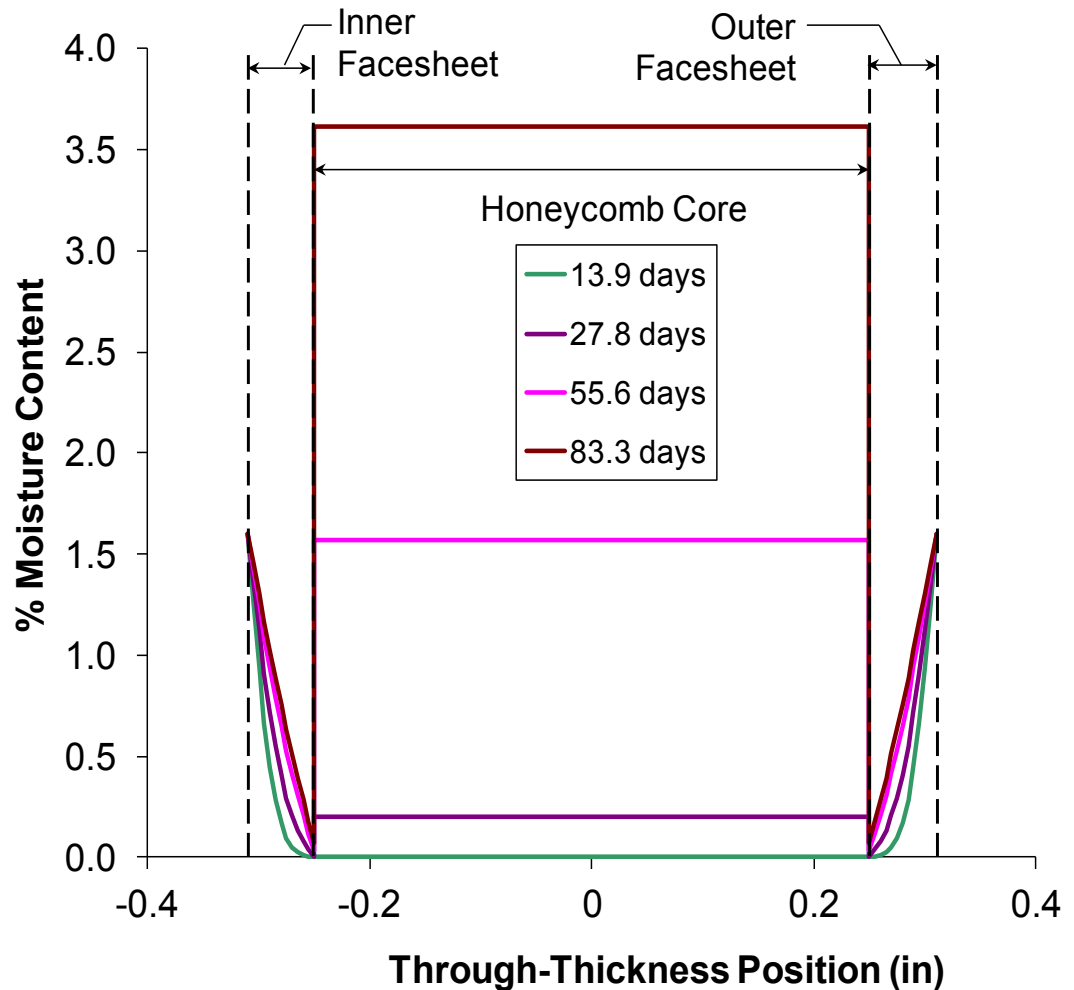
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# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

Moisture diffusion in honeycomb sandwich panel:

-12-ply Gr-Ep facesheets

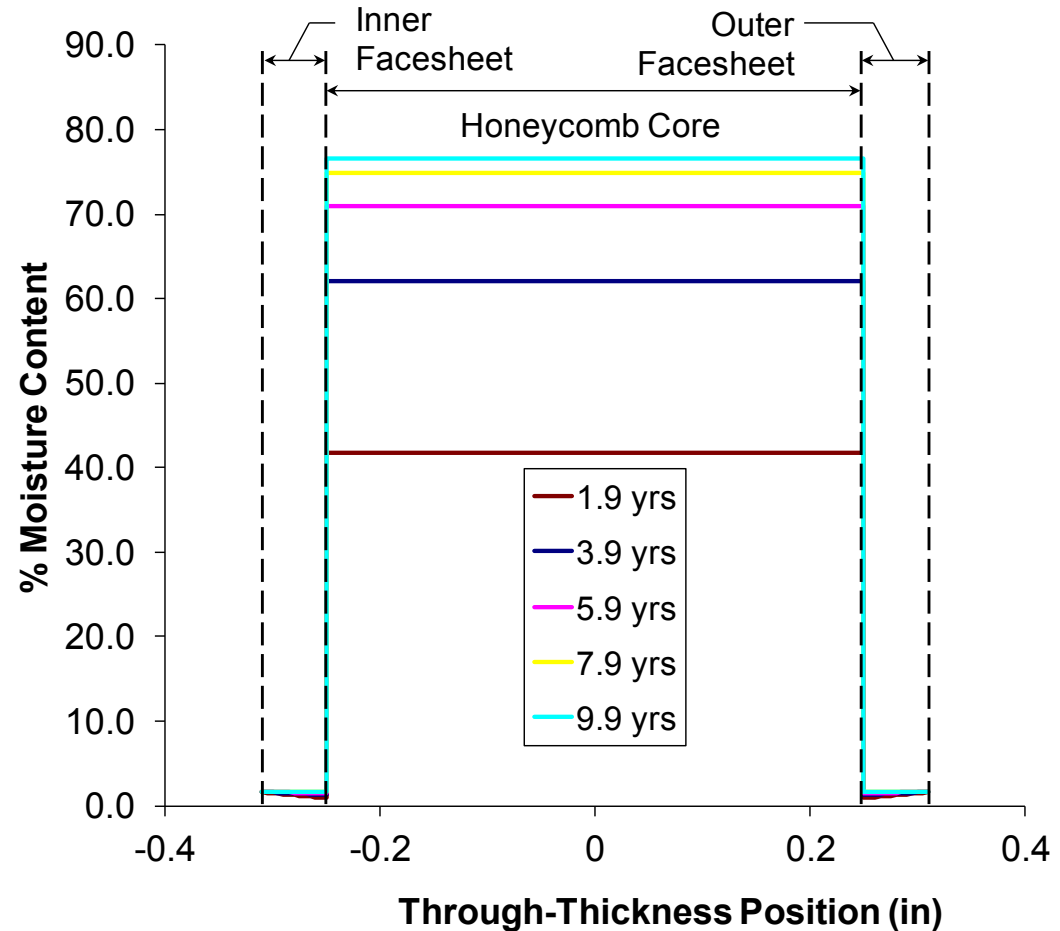
- 0.5 in Nomex core

- 90°F, 80%RH

- Core moisture content eventually equals external humidity

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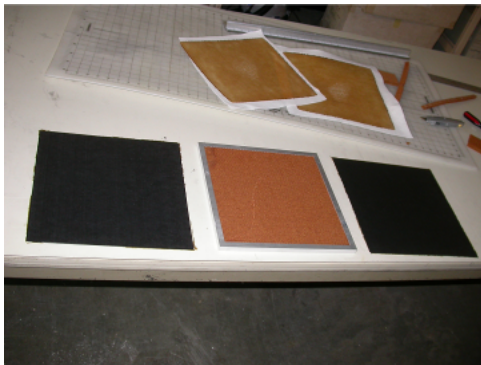
(Tuttle, AMTAS 2009)



# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

Experimental verification (Tuttle, AMTAS 2009)



- Type 410 Nomex honeycomb core
- $[0/45/90/-45]_s$  Gr/Ep facesheets
- Core sized to fit within aluminum frame to insure 1-D, through-thickness diffusion

- First facesheet bonded to one side of panel using thin-film adhesive
- Pocket for embedded humidity sensors and thermocouples milled in core





# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

Experimental verification (Tuttle, AMTAS 2009)

Sandwich panel internally instrumented with:

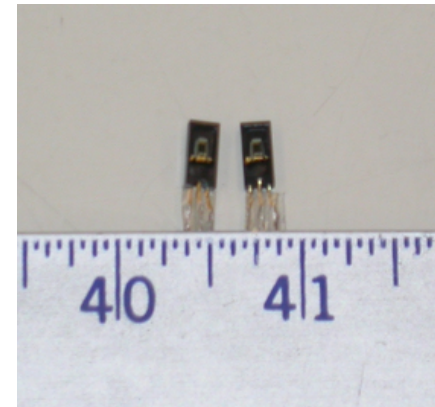
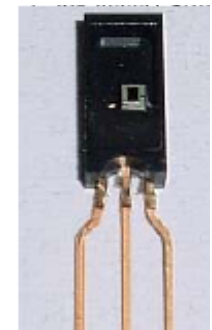
- 2 type K thermocouples
- 2 Ohmic Instruments Model HC-610 capacitive humidity sensors:
  - 5-95 %RH
  - 40 to 185°F operating range

[www.ohmicinstruments.com/](http://www.ohmicinstruments.com/)

HC-610 Thermoset polymer capacitive humidity sensor. Hybrid electronics. Linear output. Range 5 to 95 %RH 2%. Temp. - 40 to 185 °F. Supply voltage 4.0 - 5.8 VDC

PDF Data

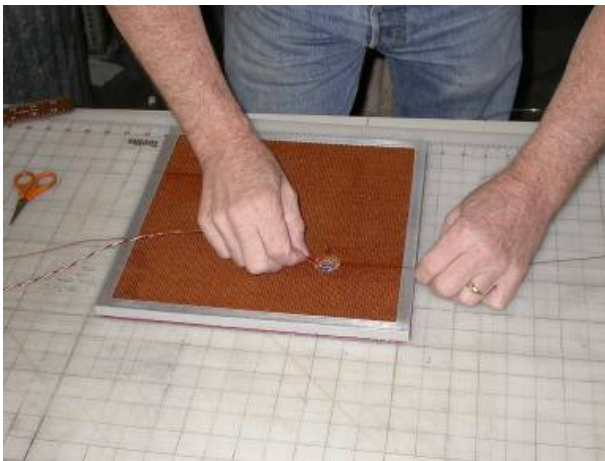
PDF Man/Instructions



# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

Experimental verification (Tuttle, AMTAS 2009)



- Leadwires inserted through honeycomb and aluminum frame

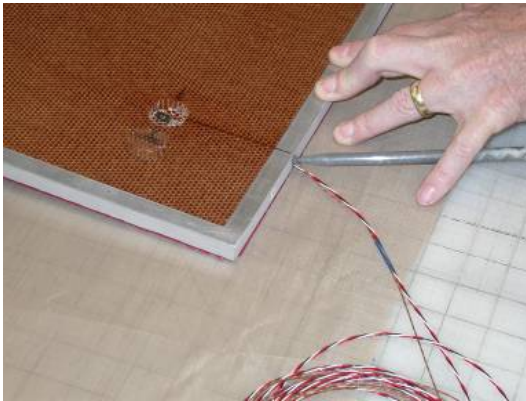
- Installation of embedded sensors



# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

Experimental verification (Tuttle, AMTAS 2009)



- Leadwire passage in aluminum frame sealed with epoxy

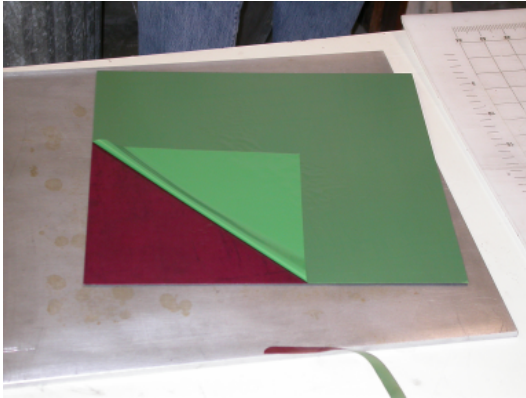
- Honeycomb 'caps' placed over instrumented sites



# Effects of Moisture Diffusion in Sandwich Composites

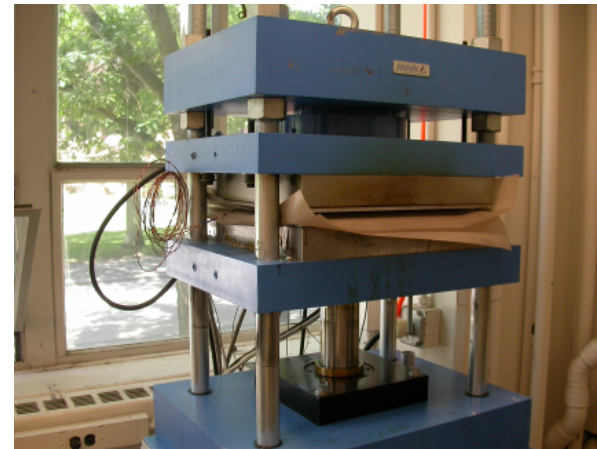
## Motivation and Key Issues:

Experimental verification (Tuttle, AMTAS 2009)



- Second facesheet bonded to panel using thin-film adhesive...

...and hot press



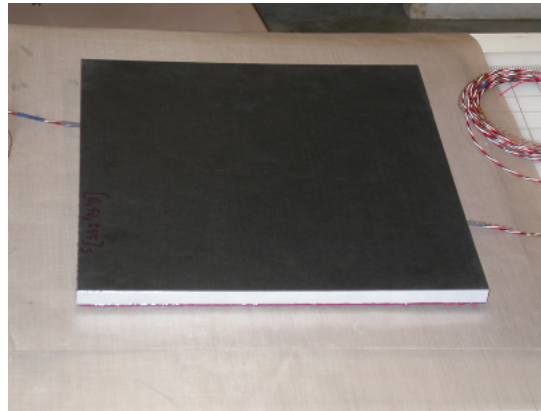


# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

Experimental verification (Tuttle, AMTAS 2009)

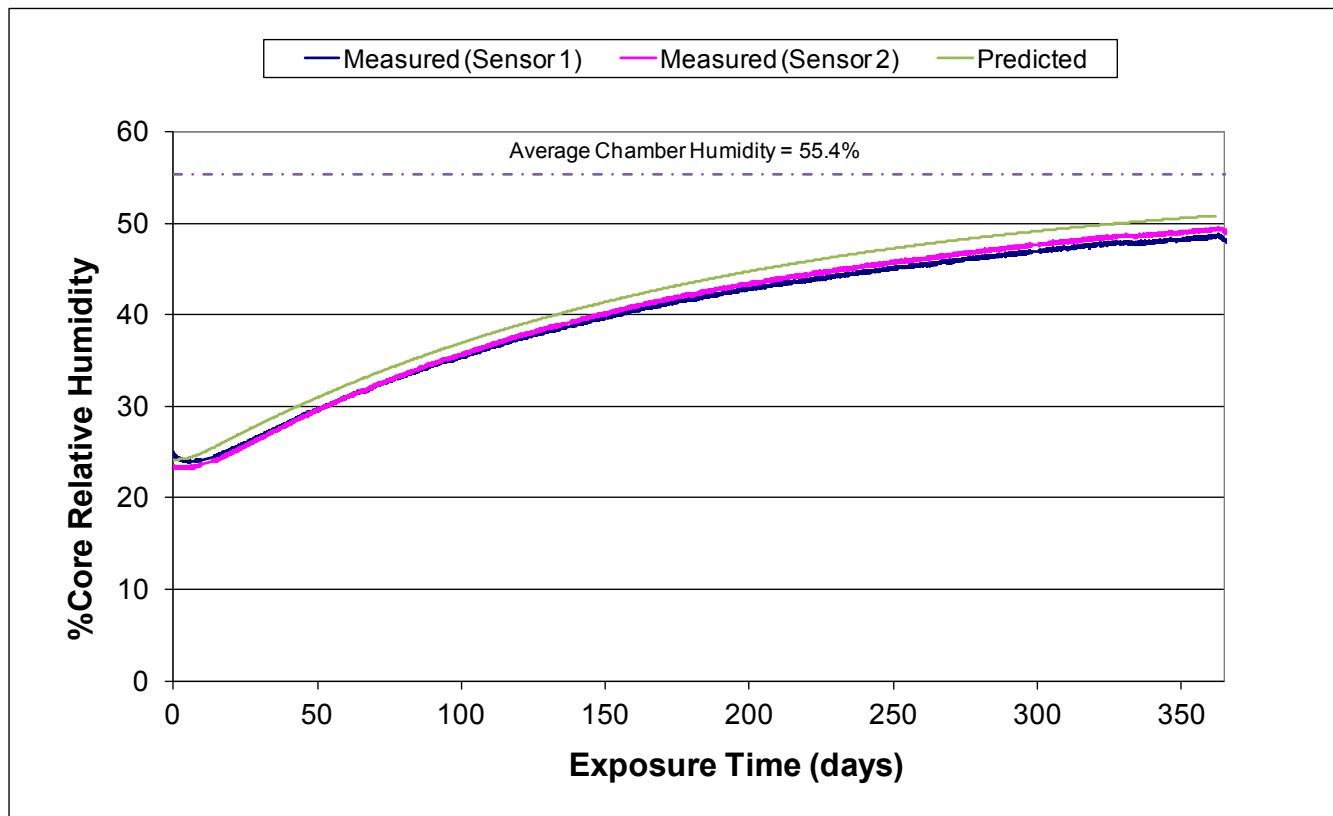
- Completed panel mounted in humidity chamber and exposed to constant environmental conditions for 12 months:
  - 40°C (104°F)
  - 55% RH
  - from 5 Aug '08 to 4 Aug '09
  - Sensors monitored continuously (i.e., every 30 minutes) using LabView



# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

Experimental verification (Tuttle, AMTAS 2009)



# Effects of Moisture Diffusion in Sandwich Composites

## Motivation and Key Issues:

- Honeycomb panels mounted on transport aircraft routinely experience pronounced thermal cycles:  
(ground level temperatures) ↔ (-60°F at 35,000 ft)
- Implication: Over long times internal core humidity will increase to a level at which a condense-freeze-thaw-evaporate cycle may occur during each flight....may represent a long-term durability issue



# Effects of Moisture Diffusion in Sandwich Composites

Objective: Determine if condense-freeze-thaw-evaporate cycle within core region cycle is detrimental. Specifically, subject representative honeycomb sandwich panels to high humidity and thermal cycles, and then measure if any

- Damage to facesheets, bondline, or core occurs (using optical microscopy)
- Change in effective bending stiffness occurs (using 4-point bending test)
- Change in strain-energy release rate  $G_{iC}$ , occurs (using single cantilever beam specimen under development by CMH-17 Sandwich Disbond Task Group)

# Effects of Moisture Diffusion in Sandwich Composites

- **Principal Investigator**
  - Mark Tuttle
- **Students**
  - William Smoot, Sung Lin 'Jason' Tien, Shuyu 'Frank' Xia
- **FAA Technical Monitor**
  - Lynn Pham
- **Industry Participation**
  - Bill Avery/The Boeing Company
  - Dan Holley and Chris Praggastis/3M
  - Bob Fagerlund/Bell Helicopter
- **Study Initiated in September 2015**



# Effects of Moisture Diffusion in Sandwich Composites

Technical Approach (some details still TBD):

- Produce 20, 2 in x 12 in specimens with 4-ply *woven* facesheets with  $[45/0/0/45]_T$  stacking sequence:
  - 5 specimens: inspect using optical microscopy and measure as-produced RT properties
  - 5 specimens: cycle as-produced panels between RT and  $-60^\circ\text{F}$ , then inspect using optical microscopy and measure RT properties
  - 10 specimens: increase core humidity to  $\sim 70\%RH$  (expect to require about 4 mos exposure time)
    - 5 specimens : inspect using optical microscopy and measure RT properties
    - 5 specimens : cycle between room temp and  $-60^\circ\text{F}$ , then measure RT properties

# Effects of Moisture Diffusion in Sandwich Composites

## Technical Approach (some details TBD):

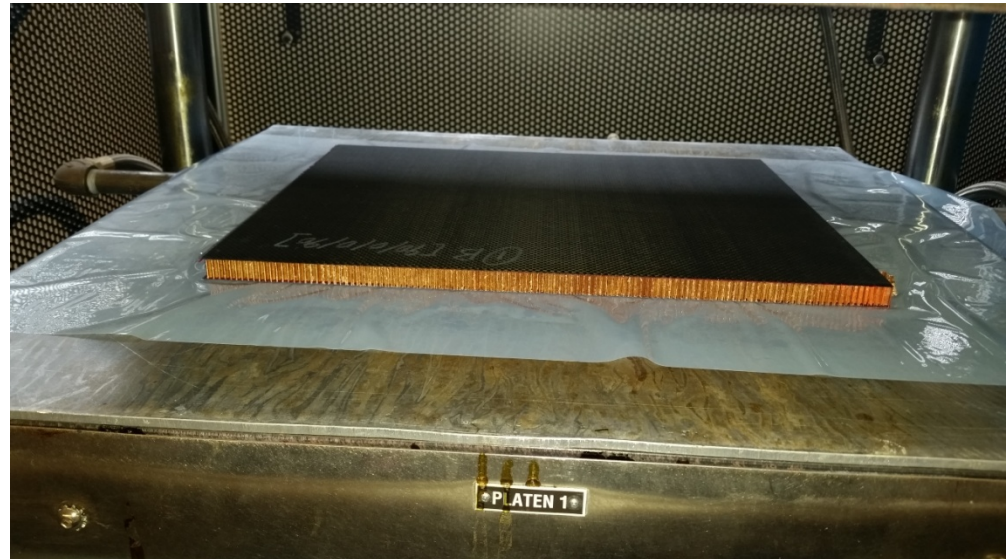
- Materials:

- Cycom 970/PWC graphite/epoxy (certified to BMS 8-256), based on:
  - Cytec 970 epoxy resin
  - Torayca T300 3K woven fabric
- Hexcel HRH-10 1/8-3.0 Nomex honeycomb core, 1/2 in thick
- 3M AF 163-2k film adhesive

- Fabrication:

- [45/0/0/45]<sub>T</sub> facesheets first produced using an autoclave cure
- Secondary bonding operation used to bond facesheets to core
- All cured materials stored in humidity chamber at 122°F and ~7%RH to minimize initial moisture content

# Effects of Moisture Diffusion in Sandwich Composites





# Effects of Moisture Diffusion in Sandwich Composites



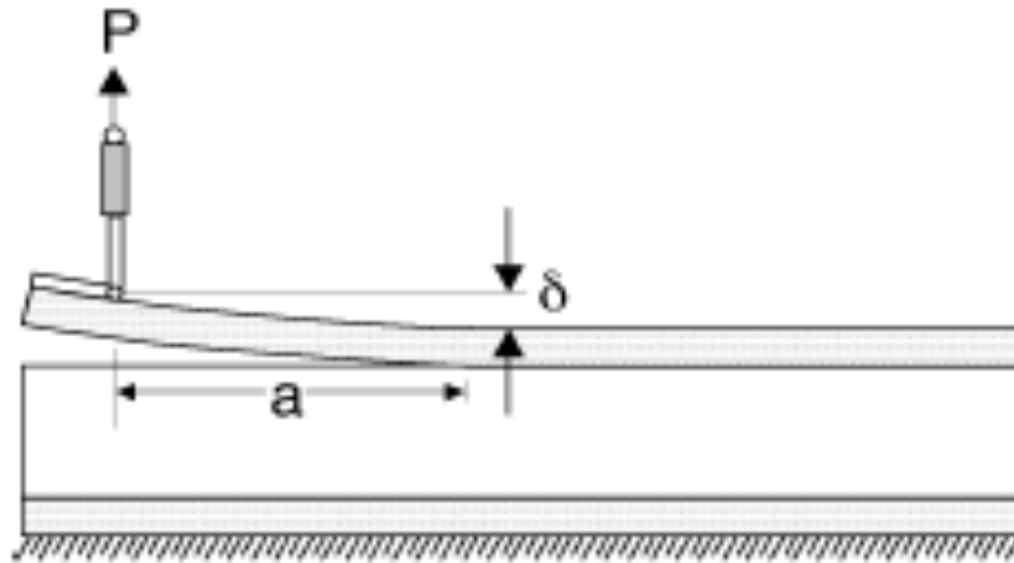
# Effects of Moisture Diffusion in Sandwich Composites

## Current Status

- Sufficient number of panels/specimens have been fabricated
- Single Cantilever Beam (SCB) test fixture nearing completion
  - Patterned after NIAR fixture
  - Similar to fixtures used by other members of CMH-17 Disbond working group
- Initial Testing to begin on/about 4 April
- Environmental conditioning to begin on/about 11 April

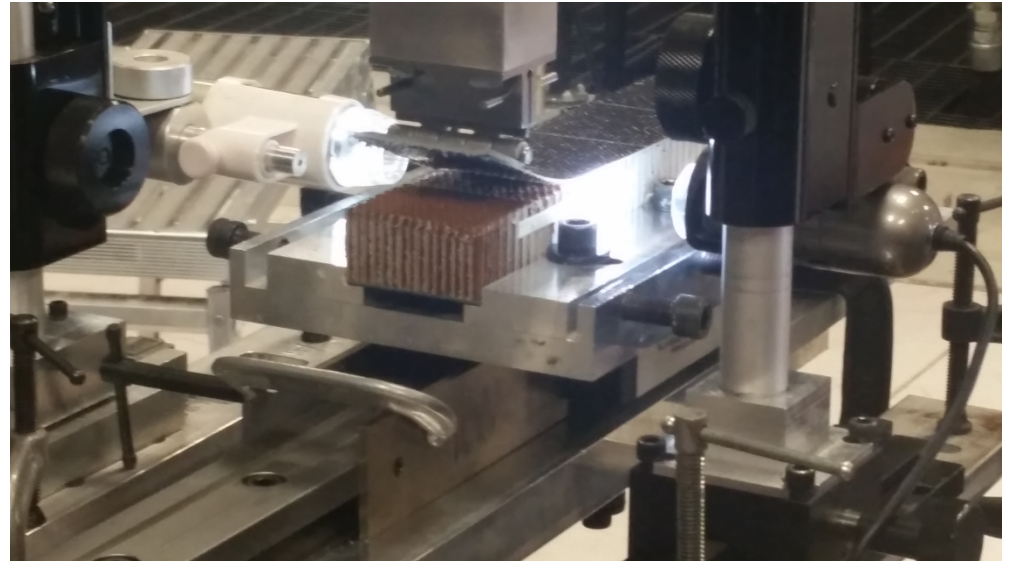
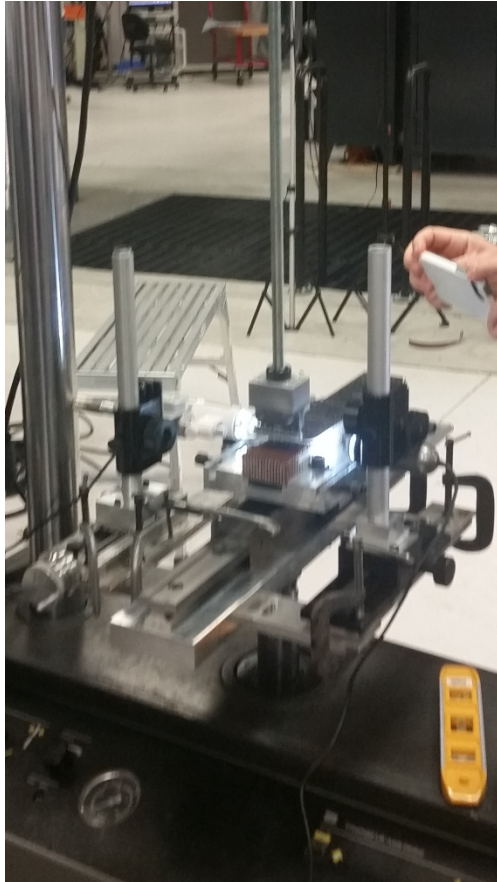


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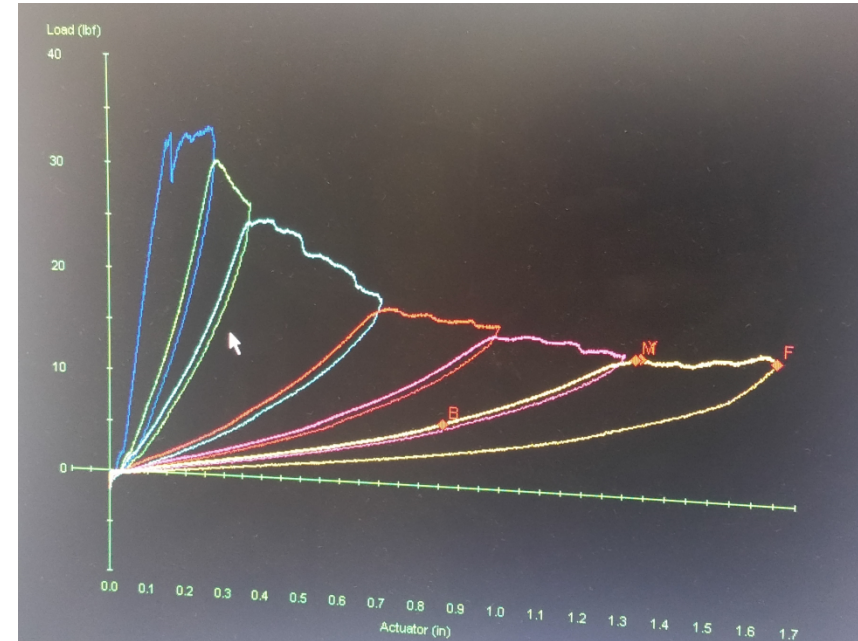
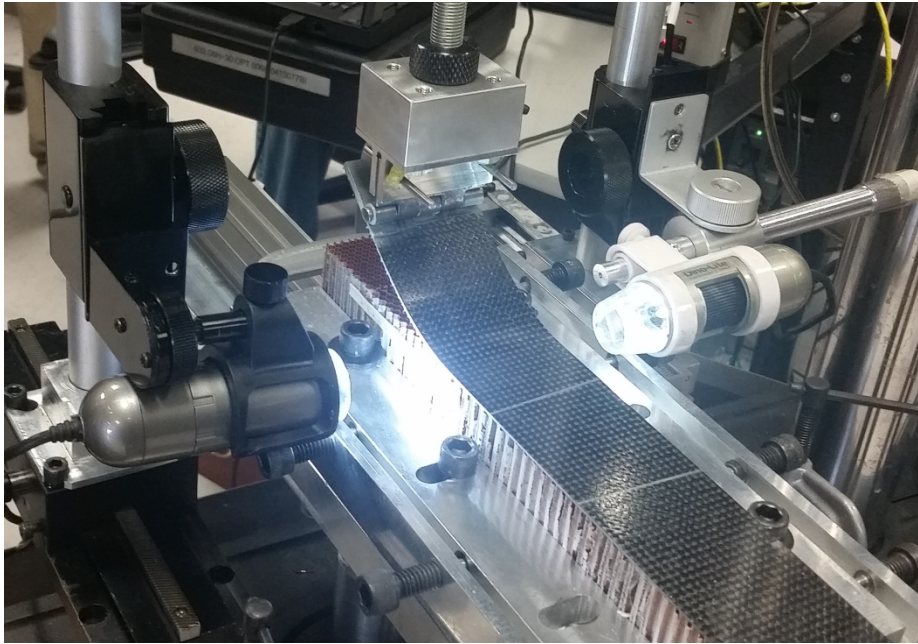
Schematic of experimental arrangement to measure  $G_1$  for sandwich panels (under development by CMH-17 working group)

# Effects of Moisture Diffusion in Sandwich Composites



Photos of test setup at NIAR

# Effects of Moisture Diffusion in Sandwich Composites



Photos of test setup at NIAR

# Effects of Moisture Diffusion in Sandwich Composites

## Benefit to Aviation:

- May identify a mechanism leading to initiation and growth of skin-core disbond in sandwich structures
- Will contribute to efforts to establish standard test protocols and data reduction practices for SCB testing of sandwich specimens

# Effects of Moisture Diffusion in Sandwich Composites

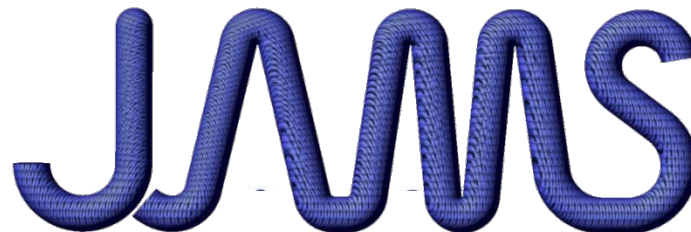
Thank You!

Questions, Comments, Suggestions?



End of Presentation.

Thank you.



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# Backup Slides





# Predicting Moisture Diffusion

- Through-thickness (1-D) diffusion of moisture assumed to be governed by Fick's first and second laws:

$$\phi = D_z \frac{\partial c}{\partial z} \qquad \frac{\partial c}{\partial t} = \frac{\partial}{\partial z} \left[ D_z \frac{\partial c}{\partial z} \right]$$

$\phi$  = rate of diffusion ("moisture flux") : units = mass/(area \* time)

$c$  = concentration : units = (mass/volume)

$D_z$  = diffusivity : units = area/time

$z$  = direction of diffusion : unit = length

$t$  = time

# Predicting Moisture Diffusion

■ From an experimental point of view it is easier to deal with percent moisture by weight ( $M$ ), rather than the concentration of moisture ( $c$ ). Fick's first and second laws are restated as:

$$\phi = \frac{D_z \rho}{100} \frac{\partial M}{\partial z} \qquad \frac{\partial M}{\partial t} = D_z \frac{\partial^2 M}{\partial z^2}$$

$\rho$  = density, mass/volume

$M$  = "moisture content"

$$M = \frac{(\text{current weight}) - (\text{dry weight})}{(\text{dry weight})} \times 100\%$$

# Predicting Moisture Diffusion

- Temperature dependency of diffusion coefficient for solids (i.e., ply and core paper) assumed to follow a Arrhenius-type relationship:

$$D = D_o \exp\left(-\frac{E}{T}\right)$$

where:  $D_o$ ,  $E$  = known material constants  
(differ for ply and core paper)  
 $T$  = absolute temperature

# Predicting Moisture Diffusion

- Temperature dependency of diffusion of H<sub>2</sub>O vapor in air assumed to follow a power law of the form\*:

$$D_{air} = 0.03376 \left( \frac{T(^{\circ}R)}{491.67(^{\circ}R)} \right)^{1.81} \frac{in^2}{sec}$$

\* Massman, W.J., *Atmospheric Environment*, Vol 32 (6), pp 1111-1127 (1998).

# Predicting Moisture Diffusion

## *Estimated Core Density and Diffusivity*

$$\rho_{core} = (V_{air})(\rho_{air}) + (V_{paper})(\rho_{paper})$$

$$D_{core} = (V_{air})(D_{air}) + (V_{paper})(D_{paper})$$

# Predicting Moisture Diffusion

The moisture content ( $M$ ) of any surface layer in contact with air can be related to the relative humidity according to (Springer, 1980):

$$M = M_u \left( \frac{\%RH}{100} \right)^b$$

- constant  $M_u$  = material property
- exponent  $b = 1$  for most materials
- relationship used to define the boundary condition at all ply interfaces



# Predicting Moisture Diffusion

Preceding relations allows forward-difference solution to Fick's equations; summary

- (At all interior ply interfaces) moisture flux leaving ply  $k$  must equal moisture flux entering ply  $k+1$
- (Boundary conditions): 
$$M = M_u \left( \frac{\%RH}{100} \right)$$
- (Initial conditions): Initial through-thickness moisture content assumed uniform (assumed = zero in '03)
- Time step increment of 1 minute

# Predicting Moisture Diffusion

## Properties Used in '03

Property	Gr/Ep (typical values)	Type 410, 2-mil Nomex (www.matweb.com)
$D_o$	0.010 $in^2/sec$ <i>(see note)</i>	0.006 $in^2/sec$
$E$	10300 °R	9000 °R
$M_u$	0.02	0.03
Density, $\rho$	0.054 $lbm/in^3$	0.026 $lbm/in^3$

Note: Properties reported for Gr/Ep vary widely. For example:

$$0.005 < D_o < 0.040 \text{ } in^2/sec$$