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CENTER OF EXCELLENCE

# Development of Environmental Durability Test Methods for Composite Bonded Joints

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# FAA Sponsored Project Information

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- **Principal Investigators:**  
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- **Graduate Student Researchers:**  
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- **FAA Technical Monitor:**  
**Ahmet Oztekin**
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**Boeing: Kay Blohowiak, Will Grace, Charles Park**  
**Air Force Research Laboratory: Jim Mazza**  
**3M Corp: Maha DeSilva**

# Outline

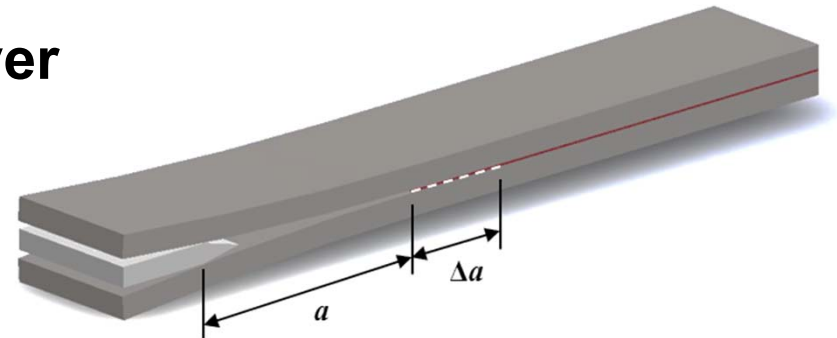
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- **Updates:**
  - Revision of metal wedge test method (ASTM D3762)
  - ASTM Adhesive Bonding Task Group D14.80.01
- **Primary focus: Environmental durability test methods for composite bonded joints**
  - Composite wedge test development
  - Comparison of results with other test methods
  - “Smart Wedge” traveling wedge test concept

# Background: The Metal Wedge Test

ASTM D3762: “Standard Test Method for Adhesive-Bonded Surface Durability of Aluminum (Wedge Test)”

- Bonded aluminum double cantilever beam loaded by forcing a wedge between adherends
- Assembly placed into test environment (ex: 50<sup>o</sup> C, 95% RH)
- Crack growth  $\Delta a$  due to environmental exposure measured following prescribed time
- Able to assess bond quality quickly by causing rapid hydration of oxide layers



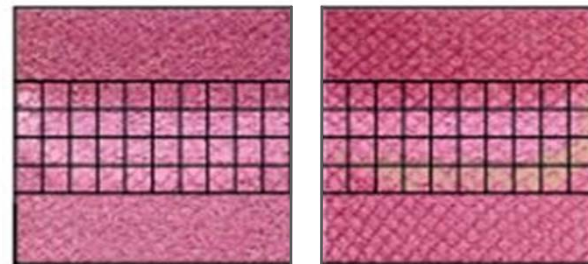
# Revision of ASTM D3762: Summary of Proposed Revisions

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- Correction of existing errors in standard
- Broadening of scope to include metals other than aluminum as adherends
- Provided additional guidance in specimen manufacturing
- Provided additional detail in test procedure
- Addition of requirement to estimate % cohesion failure in region of environmental crack growth

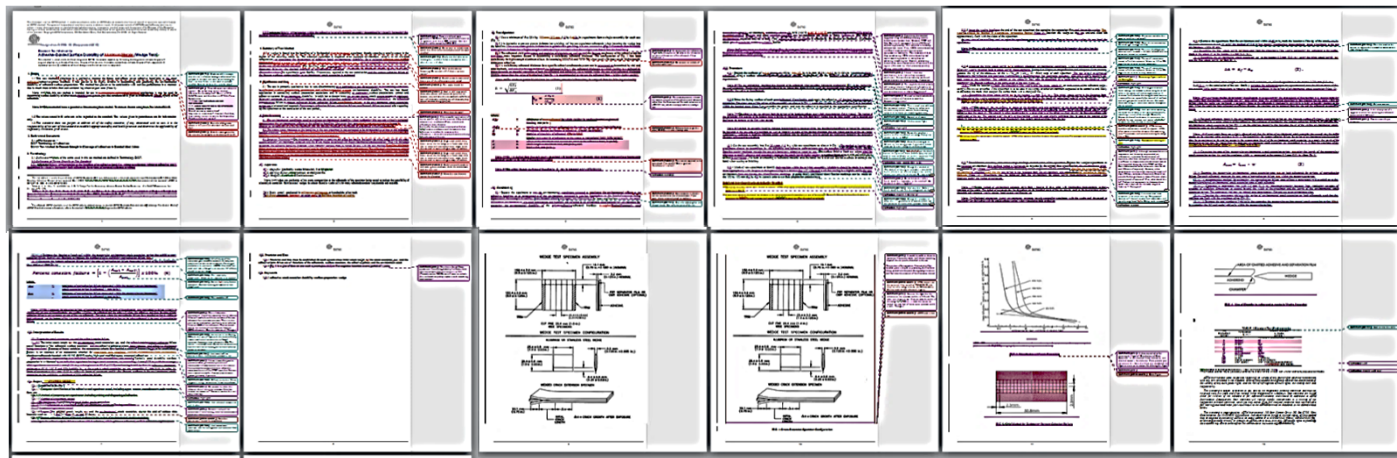
*Percent cohesion failure:*

$$\left[ 1 - \left( \frac{A_{nc1} + A_{nc2}}{A_{ext}} \right) \right] \times 100\%$$



# Revision of ASTM D3762: Current Status

- Completed extensive revision of standard
- Initial D14.80 subcommittee balloting
- Addressing remaining concerns from negative votes
- Reballot at concurrent subcommittee/main levels later this summer



# Collaborations with ASTM D14 (Adhesives): D14.80.01 Task Group



- Includes ASTM D14 (Adhesives) and ASTM D30 (Composites) members
- Meets concurrently with ASTM D30 to allow for greater participation
- Balloting through D14.80 subcommittee and D14 main
- Technical contact(s) from D30 to attend D14 meetings and provide TG status reports

## Current Activities

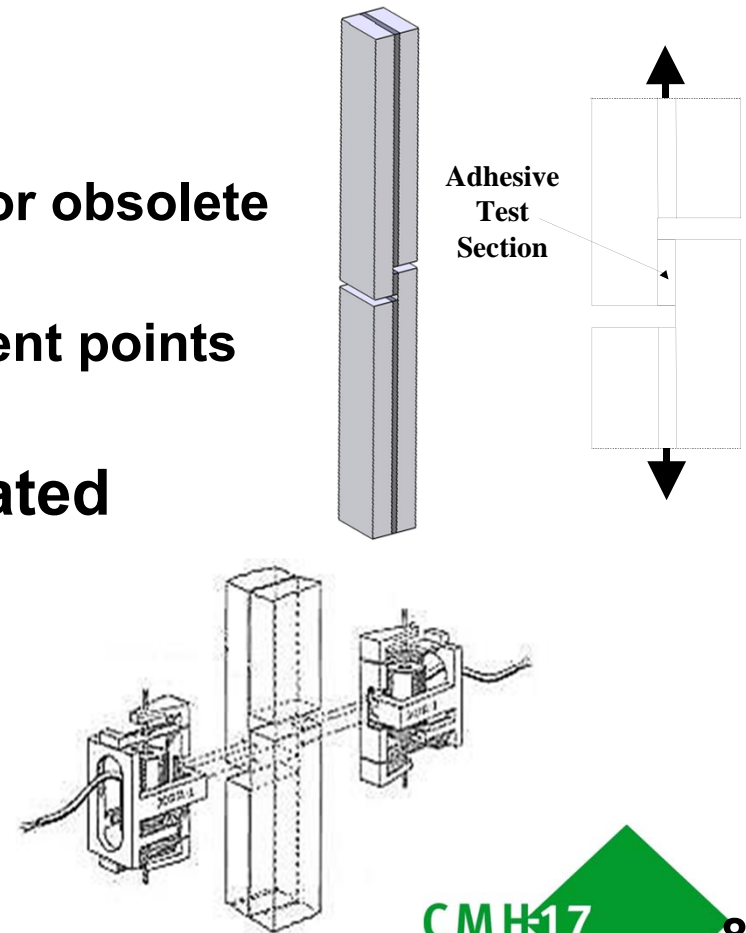
- ASTM D3762 Metal Wedge Test revision
- ASTM D5656 Thick Adherend Lap Shear Test revision
- Composite Wedge Test development/standardization

# Current Activities: ASTM D14.80.01 Task Group



## Improvements to ASTM D5656–Thick Adherend Lap Shear Test

- **Best practices for shear strain measurement**
  - Identify suitable replacement(s) for obsolete KGR-1 extensometer
  - Determination of optimal attachment points for shear strain measurement
- **Round-robin investigation initiated**
  - Two paste adhesives
  - Four adhesive thicknesses
  - Three labs/measurement methods
- **In conjunction with CMH-17 Testing Working Group**





# Outline

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  - ASTM Adhesive Bonding Task Group D14.80.01
- ➔ **Primary focus: Environmental durability test methods for composite bonded joints**
  - Composite wedge test development
  - Comparison of results with other test methods
  - “Smart Wedge” traveling wedge test concept

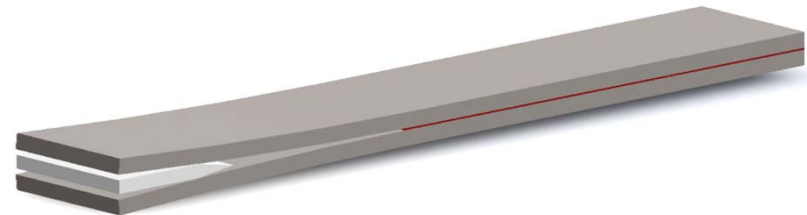
# Overview:

## Development of a Composite Wedge Test:

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### Additional Complexities:

- Variable flexural rigidity ( $E_f \cdot I$ ) of composite adherends
- Environmental crack growth dependent on adherend flexural rigidity
  - Flexural rigidity must be within an acceptable range  
or...
  - Must tailor wedge thickness for composite adherends  
or...
  - Must use another quantity to assess durability



# Why Environmental Durability Tests of Composite Bonded Joints?

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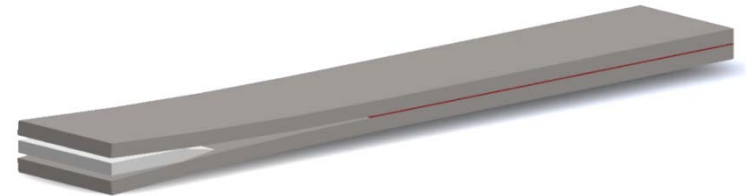
*“There is currently no known mechanism similar to metal-bond hydration for composites”*

- **Ensure longer-term environmental durability of composite bonds**
- **Investigate effects of environmental exposure on performance of bonded composite joints**
  - Failure mode: cohesion versus adhesion failure
  - Estimate fracture toughness reduction
- **Evaluate effectiveness of surface preparation**

# Use of Fracture Toughness, $G_c$ To Assess Environmental Durability

Consider composite adherends as cantilever beams

- Measured values of crack length,  $a$
  - Known value of beam deflection,  $\delta$
- $\delta = t/2$  (half of wedge thickness)



Tip deflection of a cantilever beam:  $\delta = \frac{t}{2} = \frac{P l^3}{3 E_f I} = \frac{T a^3}{3 E_f I}$

$$T = \frac{E_f b h^3 t}{8 a^3}$$

Strain energy due to bending:  $U = \frac{1}{2} T \delta$

Strain energy release rate:  $G_c = \frac{dU}{dA}$

$$\Rightarrow G_c = \frac{3E_f t^2 h^3}{16 a^4} \left[ \frac{1}{\left(1 + 0.64 \frac{h}{a}\right)^4} \right]$$

Correction factor for crack tip rotation

$a$  = crack length

$t$  = wedge thickness

$h$  = adherend thickness

$b$  = specimen width

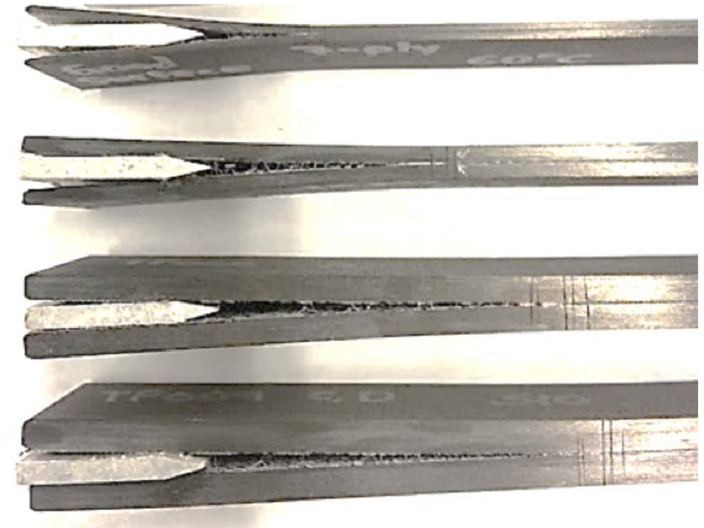
$T$  = load to deflect tip of beam

$E_f$  = flexural modulus

$G_c$  = fracture toughness

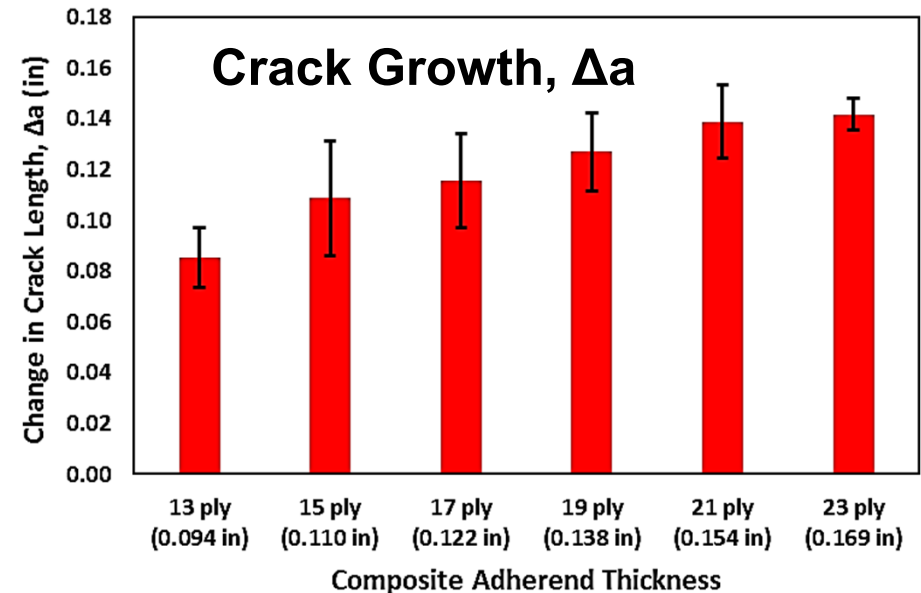
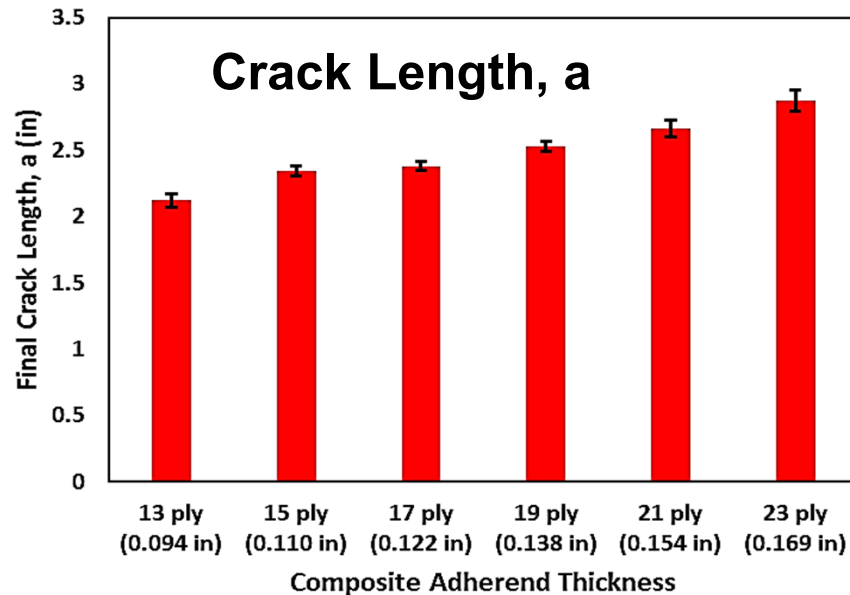
# Experimental Investigation: Composite Wedge Test Development

- IM7/8552 carbon/epoxy adherends
- AF163-2K film adhesive
- “Ideal Bond”: Grit-blast & acetone wipe bond surfaces
- Multiple adherend thicknesses to produce different flexural rigidities ( $E_f * I$ )
  - 13, 15, 17, 19, 21, 23 ply thicknesses
  - (0.10 to 0.17 in thick adherends)
- 122°F (50°C) and 95% humidity environment for 5 days



# Effects of Composite Adherend Thickness: Crack Length and Growth Measurements

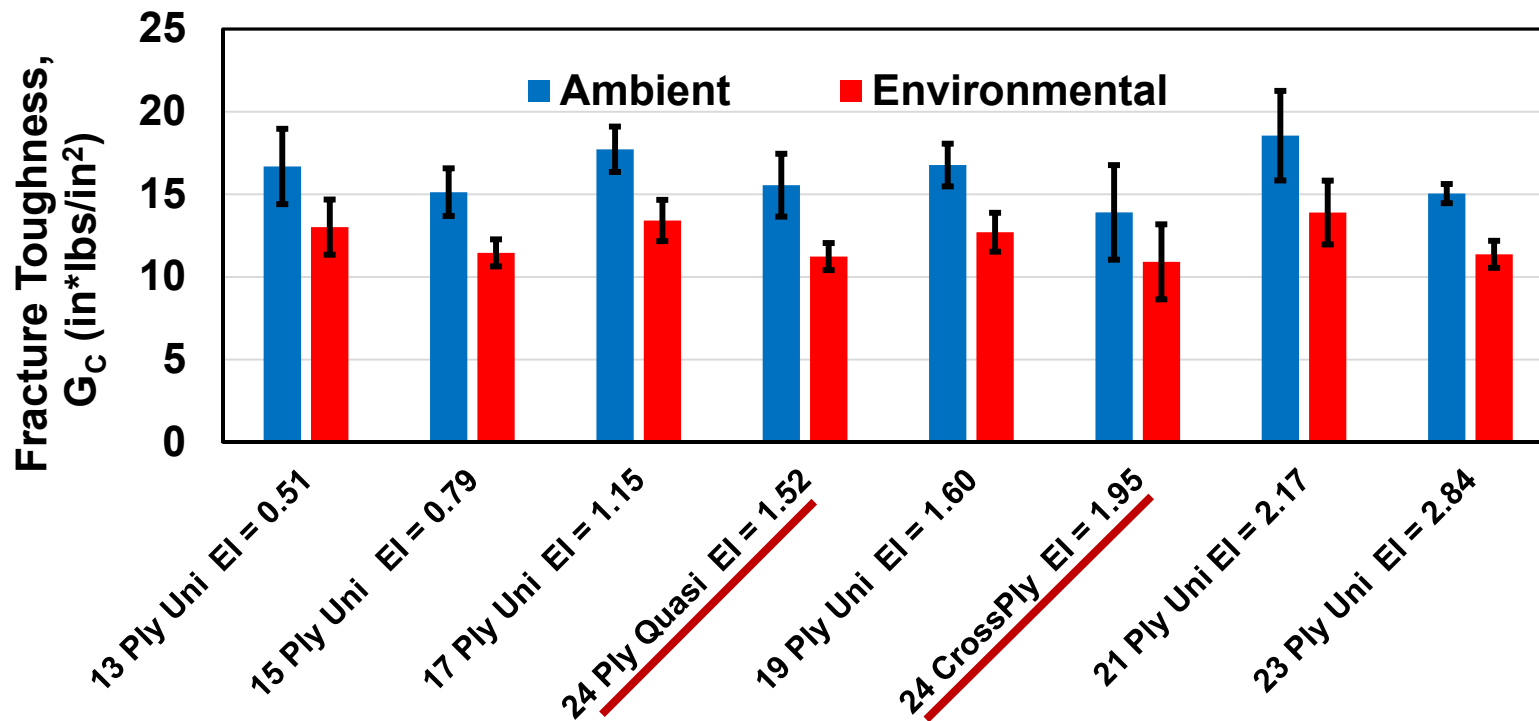
122°F (50°C) and 95% humidity environment



**Increasing adherend thickness (and flexural stiffness)...**

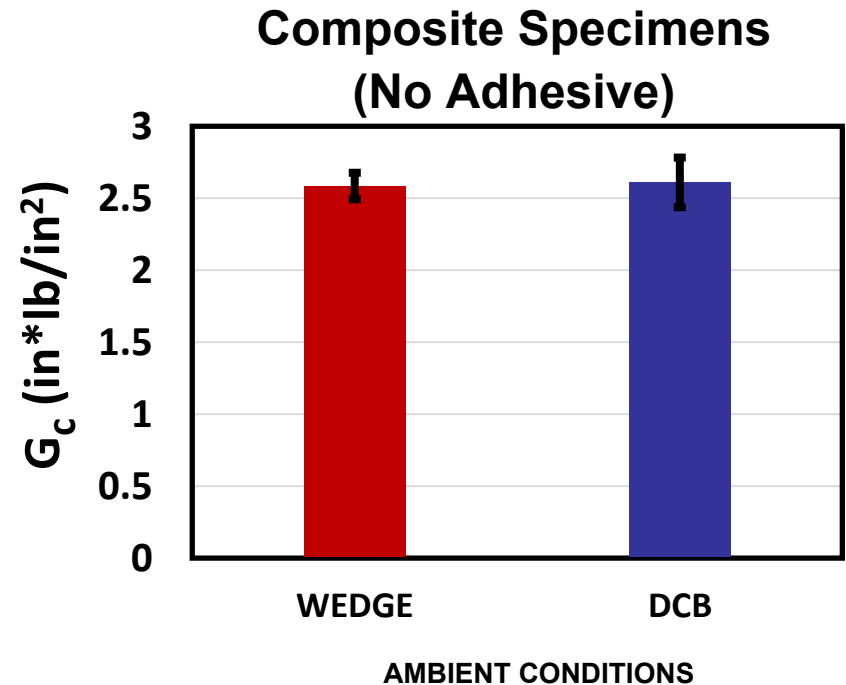
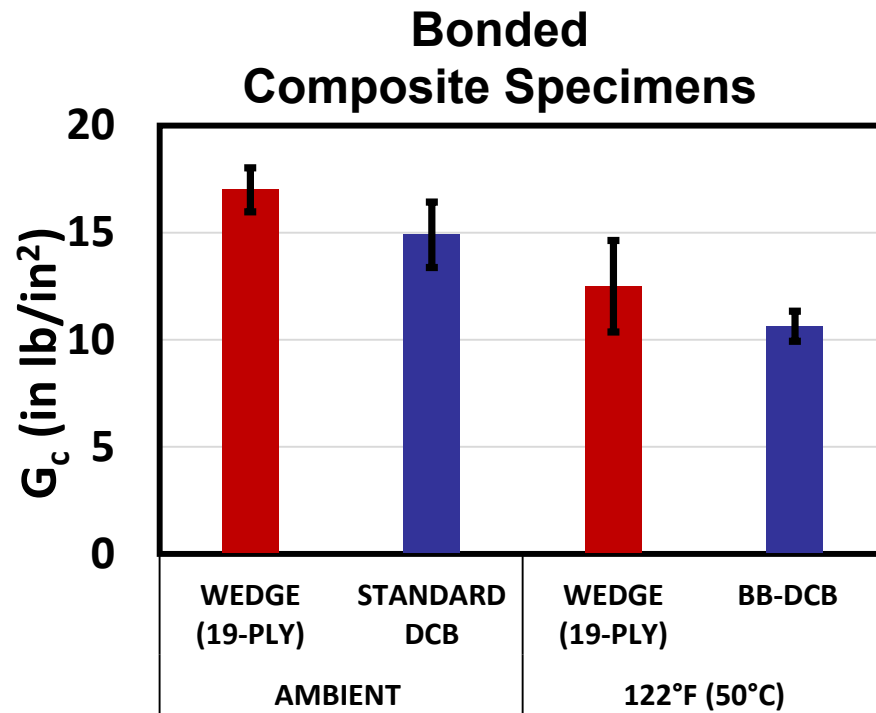
- Increases crack length, a
- ➔** Increases crack growth,  $\Delta a$

# Wedge Testing of Multidirectional Laminates: Fracture Toughness Values



- Apparent fracture toughness values remain relatively constant
- Provides estimate of fracture toughness at ambient conditions
- $G_c$  values from quasi-isotropic and crossply laminates consistent with previous unidirectional laminates

# Composite Wedge Test Development: Comparisons With DCB Test



***General agreement to date between double cantilever beam (DCB) and composite wedge test results***



## Investigating Accuracy of $G_c$ Values From Wedge Test: Determination of Flexural Modulus, $E_f$

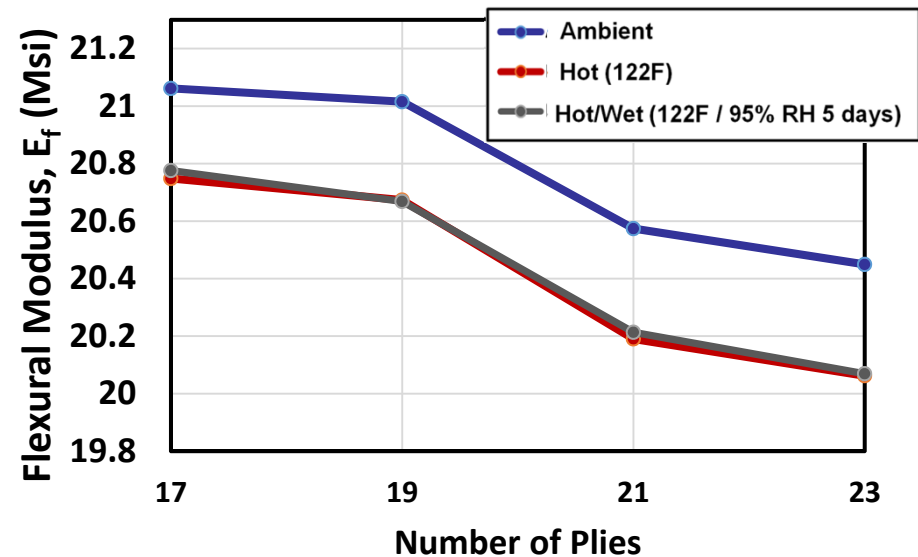
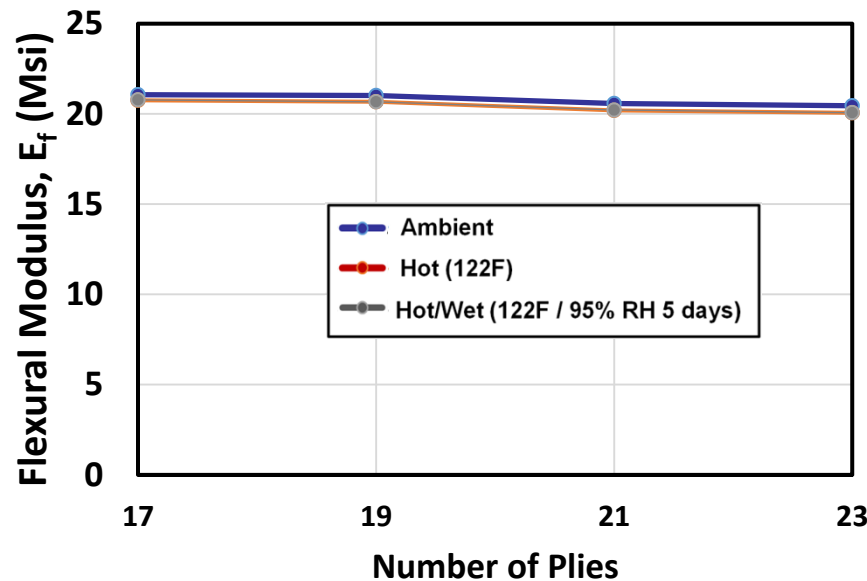
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- Require value of flexural modulus,  $E_f$ , for calculating fracture toughness,

$$G_c = \frac{3 E_f t^2 h^3}{16 a^4} \left[ \frac{1}{(1+0.64 \frac{h}{a})^4} \right]$$

- $E_f$  value should be representative of that experienced during wedge testing
- What type of test should be used to measure  $E_f$ ?
- What environment should testing be performed at:
  - *How does environmental exposure affect  $E_f$ ?*
  - *Can RT/ambient  $E_f$  measurement be used?*

## Three-Point Flexural Modulus ( $E_f$ ) Testing of Composite Adherends: Effects of Environmental Conditioning



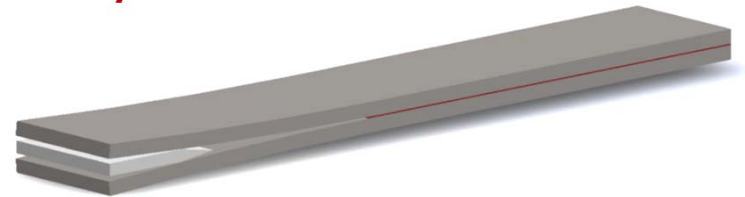
- **Three-point flexure testing**
- **Less than 2% reduction in  $E_f$  due to conditioning environment (122 °F, 95% RH for 5 days)**
- ***Flexure testing of adherends at RT/Ambient conditions appears suitable for  $E_f$  determination***

Since We're Testing Adherends to Measure  $E_f$ ...

## Why Not Measure $E_f * I$ ?

$G_c$  written in terms of flexural modulus,  $E_f$

$$G_c = \frac{3 E_f t^2 h^3}{16 a^4}$$



$G_c$  written in terms of flexural rigidity,  $E_f I$

$$G_c = \frac{9 (E_f I) t^2}{4 b a^4}$$

a = crack length  
t = wedge thickness  
h = adherend thickness  
b = specimen width  
 $E_f$  = flexural modulus  
I = area moment of inertia  
 $G_c$  = fracture toughness

Measuring flexural rigidity  $E_f I$ ...

- Allows for direct slope measurement from load/displacement curve ( $P/\delta$ )
- Eliminates need for adherend thickness measurement
- Possible elimination of correction factor

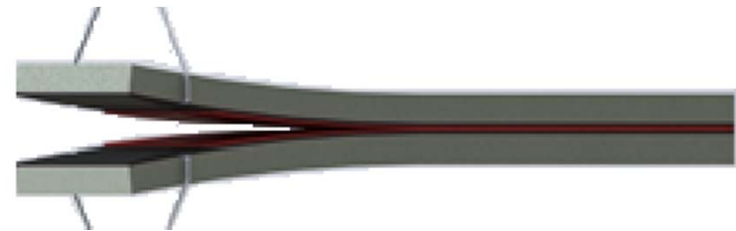
# Use of Effective Flexural Rigidity For Fracture Toughness Determination

- Express fracture toughness written in terms of  $E_f I$ :

$$G_c = \frac{9(E_f I) t^2}{4b a^4}$$

- Measure  $E_f I$  directly using post-tested wedge specimen under DCB type loading

$$E_f I = \frac{2a^3}{3} \left( \frac{\Delta P}{\Delta \delta} \right)$$



$a$  = beam span (crack length)

$P$  = applied force

$\delta$  = crosshead displacement

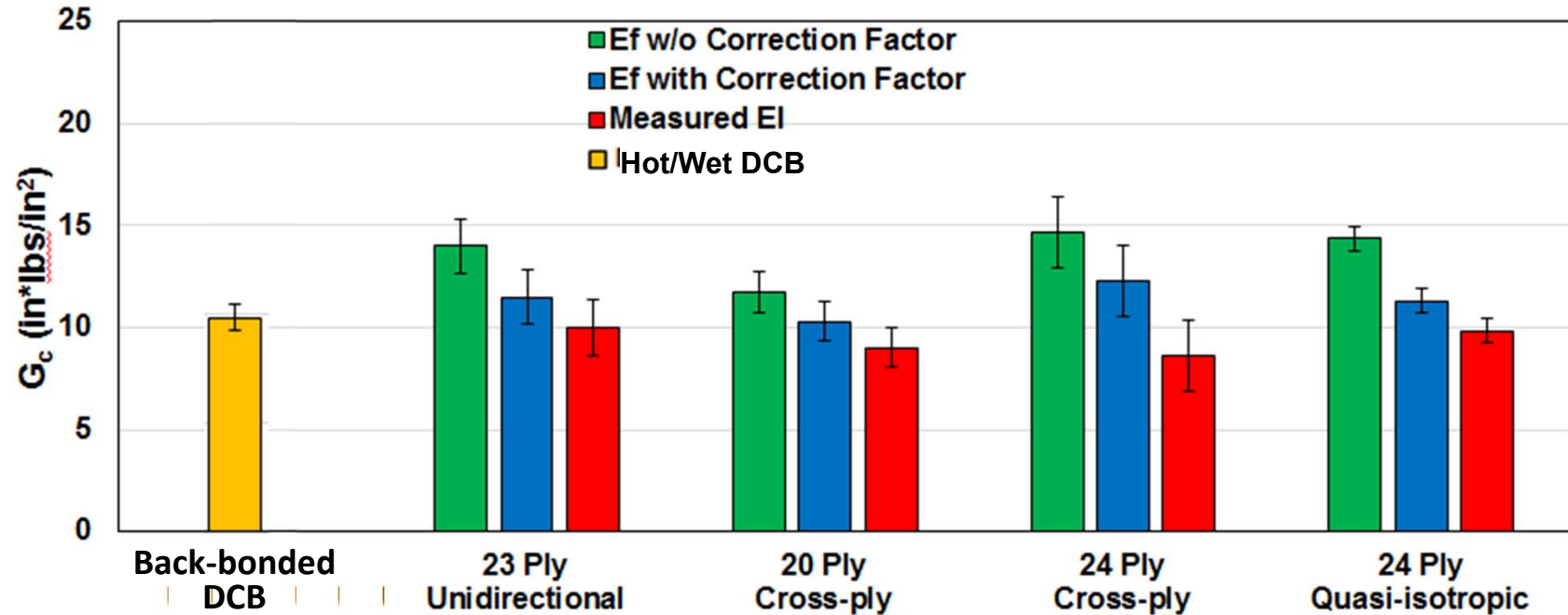
$t$  = wedge thickness

$E_f$  = flexural modulus

$I$  = moment of inertia

- Correction for crack tip rotation  
“built-in” to effective  $E_f I$  measurement

## Comparison of Wedge Test and DCB Test Results: 50°C, 95% RH, 5 days

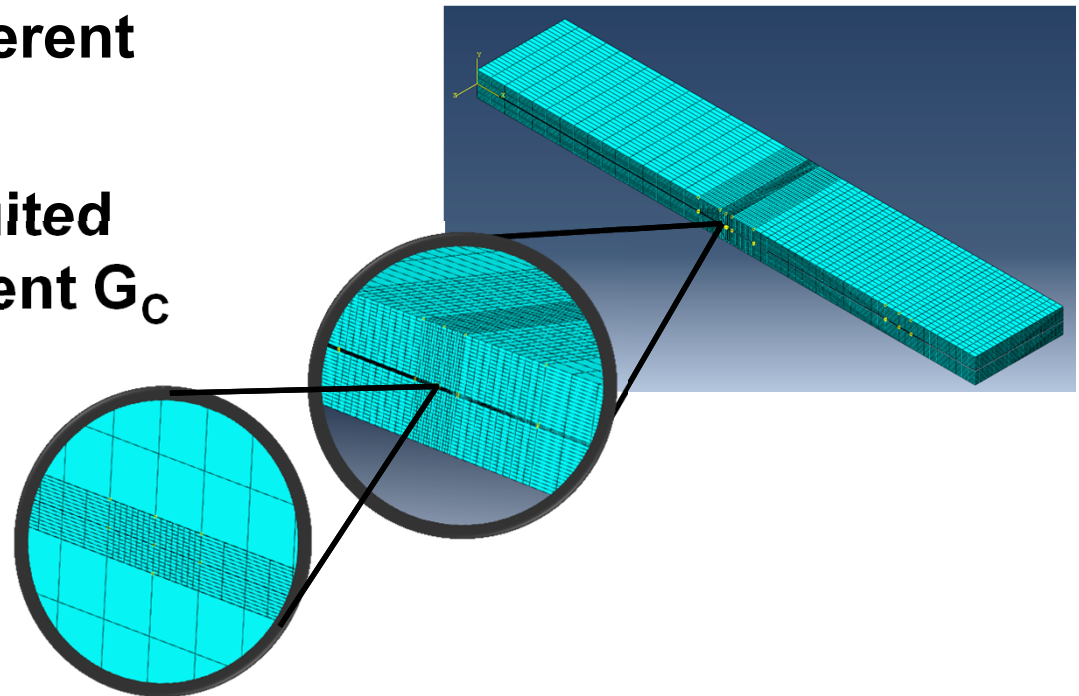


***General agreement with both closed-form correction factor and measured  $E_f$  approaches***

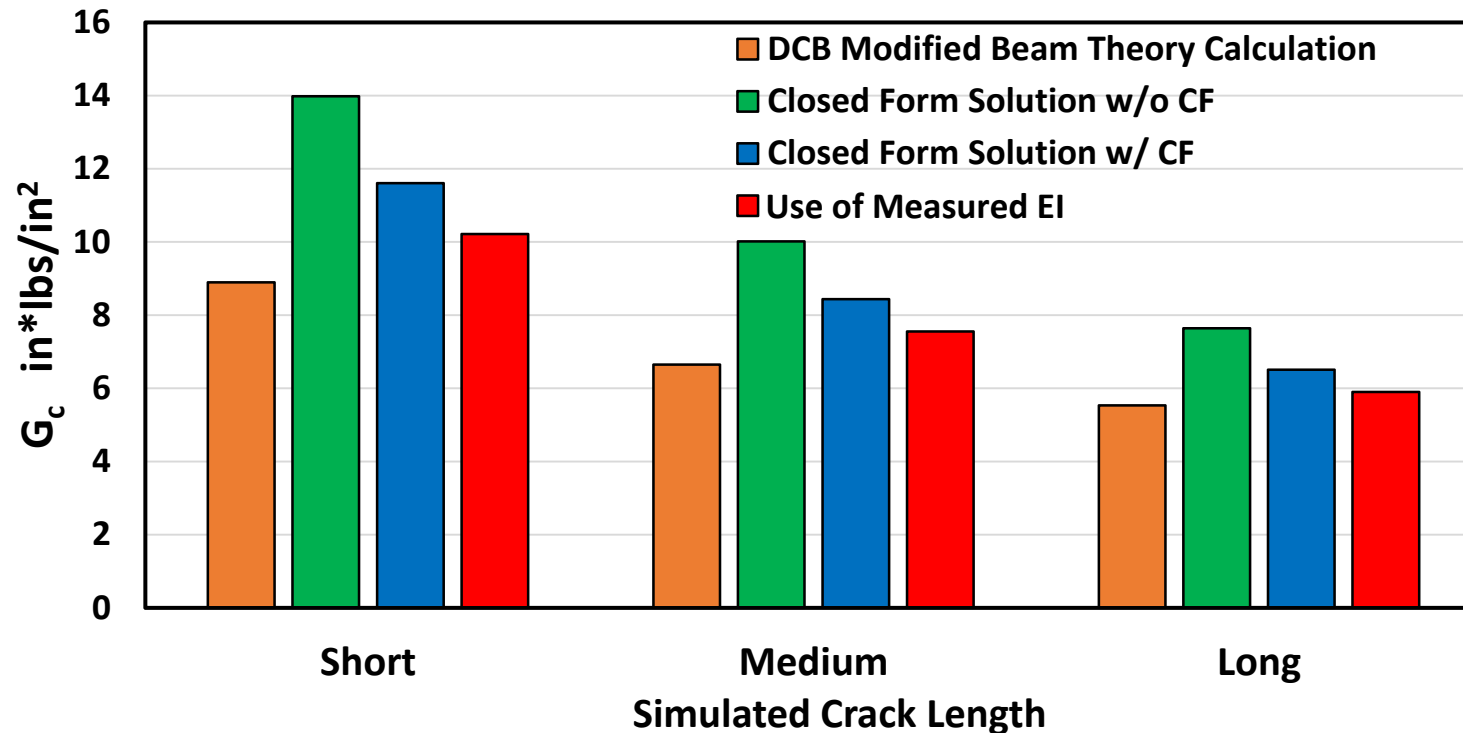
## Current Focus: Numerical Simulation To Investigate Correction Factor

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- 3D finite element analysis of composite adherends with adhesive layer
- Prescribed displacement loading simulating wedge
- Simulation of three different crack lengths
- Simulation of three different adhesive thicknesses
- Identification of best suited methodology for apparent  $G_C$  determination



# Comparison of Methods for $G_c$ Determination: Preliminary Results From Numerical Simulation



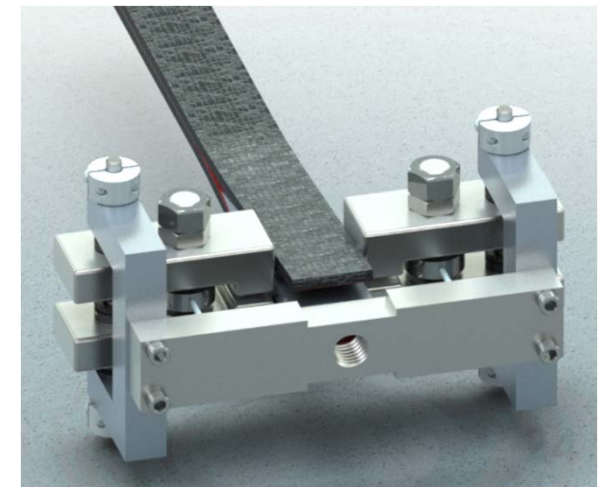
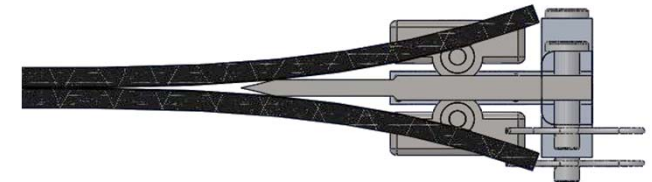
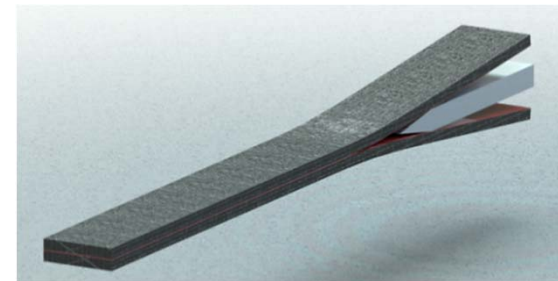
$$G_c = \frac{3 E_f t^2 h^3}{16 a^4}$$

$$G_c = \frac{3 E_f t^2 h^3}{16 a^4} \left[ \frac{1}{(1+0.64 \frac{h}{a})^4} \right]$$

$$G_c = \frac{9(E_f I) t^2}{4b a^4}$$

# *What if the Wedge Could Measure Opening Force During Wedge Testing?*

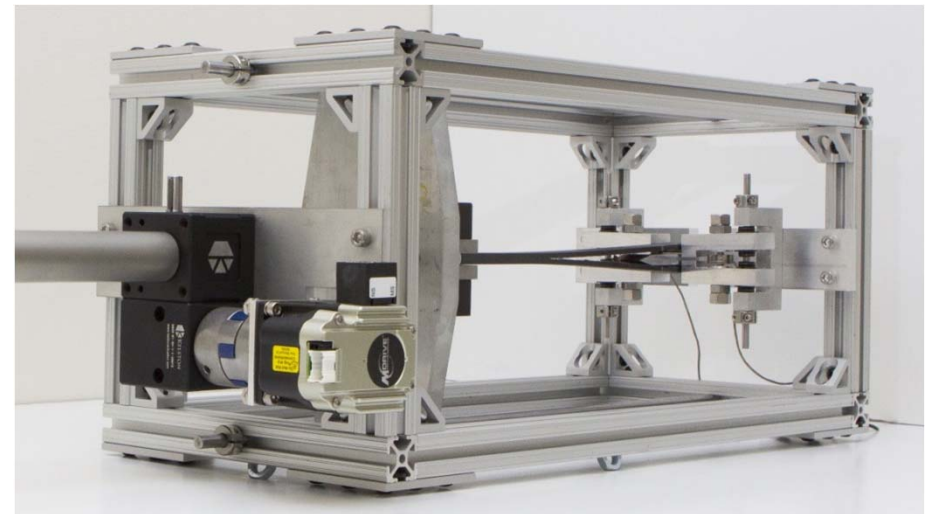
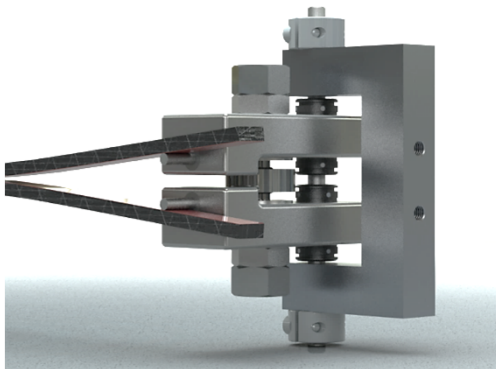
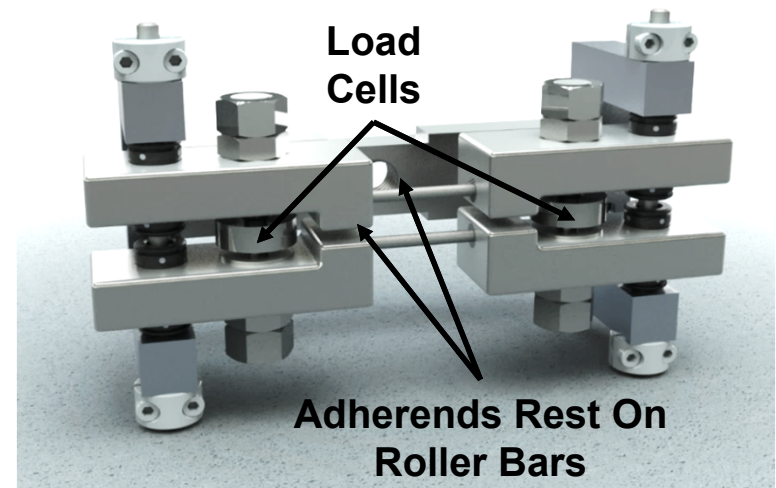
- **Continuous opening-force measurement as wedge driven through specimen**
- **Monitor for drop in measured force**
  - Increased crack length ahead of wedge
  - Reduced fracture toughness
- **Retain wedge in specimen for environmental durability test**
- **Similar to traveling wedge test, but with force-sensing “smart” wedge**





# “Smart Wedge” Concept

- Two compression load cells to measure opening force
- Adherends supported by roller bars
- Linear bearings allow for vertical displacement
- Wedge driven through bondline or held in place



## “Smart Wedge” Concept: Fracture Toughness Measurement

- $G_c$  written in terms of  $E_f I$ :  $G_c = \frac{9(E_f I) t^2}{4 b a^4}$

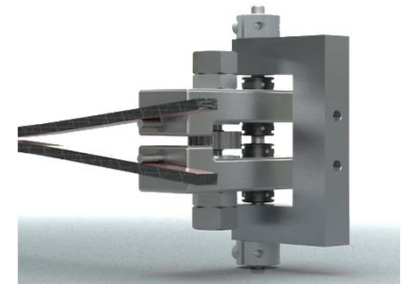
- From beam theory, solving for crack length,  $a = \sqrt[3]{\frac{3(E_f I) t}{P}}$

$$G_c = \left[ \frac{9 P^4 t^2}{4 b^3 (E_f I)} \right]^{1/3}$$

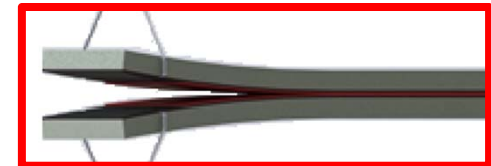
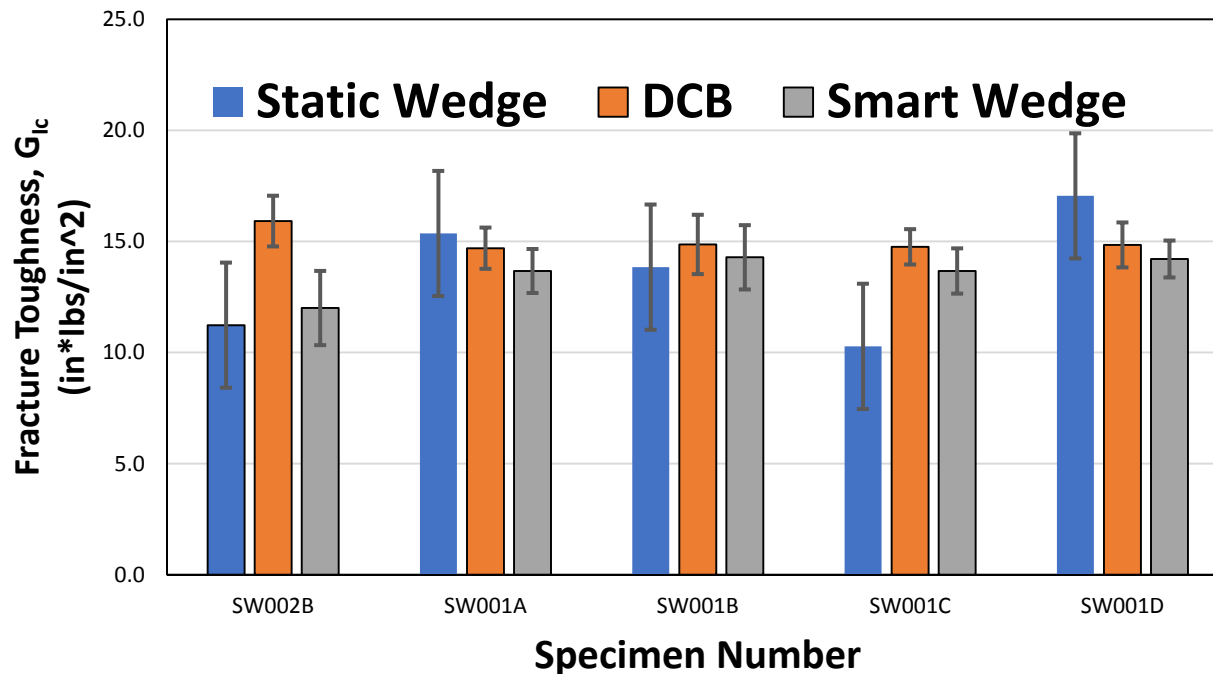
- Can calculate  $G_c$  knowing:
  - $P$  (measured force)
  - $t$  (wedge thickness)
  - Flexural rigidity,  $E_f I$  (measured)

$a$  = beam span (crack length)  
 $P$  = applied force  
 $t$  = wedge thickness  
 $E_f$  = flexural modulus  
 $I$  = moment of inertia

***Do not need crack length measurement!***



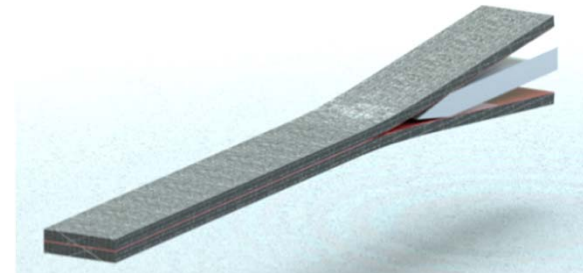
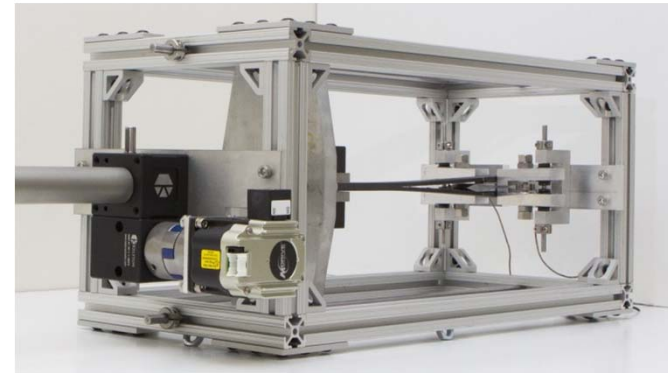
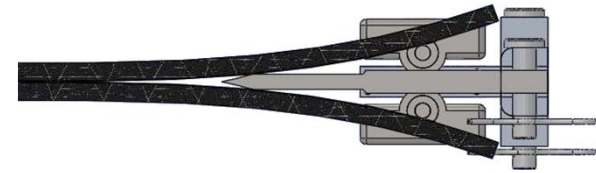
# Preliminary Results: “Smart Wedge” Testing



- 3 tests (DCB, static wedge and Smart Wedge) performed on each bonded composite specimen
- Initial results appear promising

# Smart Wedge Testing: Envisioned “Hybrid” Procedure

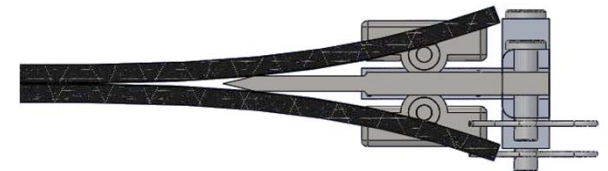
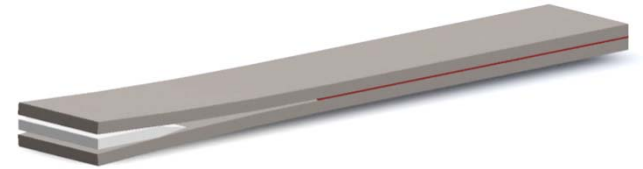
- Install specimen into smart wedge, retract installation wedge
- Obtain initial load and crack length measurements, calculate flexural rigidity,  $E_f I$
- Perform “traveling wedge” type testing, obtain real-time fracture toughness  $G_C$  estimates
- Halt traveling wedge, reinsert wedge, and subject to environmental exposure for durability assessment



Plans for Upcoming Research:  
**Development of Environmental Durability Test Methods  
for Composite Bonded Joints**

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- **Establish limits of applicability and recommended procedures for the composite wedge test**
  - **Acceptable range of flexural rigidities  $E_f I$**
  - **Method of  $G_c$  determination**
- **Assessment of “Smart Wedge concept for hybrid assessment of bond quality over larger bond areas**
  - **Continuous fracture toughness measurements**
  - **Environmental durability assessment when desired**



# BENEFITS TO AVIATION

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- **Improved environmental durability test method for metal bonds (metal wedge test, ASTM D3762)**
- **Composite wedge test for assessing the environmental durability of composite bonds and assessing surface preparations**
- **Hybrid traveling wedge/static wedge test for evaluation of larger bond areas**
- **Dissemination of research results through FAA technical reports and conference/journal publications**

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**Thank you for your attention!**

***Questions?***