DEVELOPMENT AND EVALUATION OF FRACTURE MECHANICS TEST METHODS FOR SANDWICH COMPOSITES

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Research Objectives

Develop fracture mechanics test methods for sandwich composites

- Focus on facesheet/core delamination
- Both Mode I and Mode II

Advanced Materials in Transport Aircraft Structures

Suitable for ASTM standardization





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SELECTED MODE I CONFIGURATION: Plate-Supported Single Cantilever Beam (SCB)

- Elimination of bending of sandwich specimen
- Minimal Mode II component (less than 5%)
- No crack "kinking" observed
- Appears to be suitable for a standard test method







Current Prototype Mode I Test Fixture: Single Cantilever Beam (SCB)

- Translating fixture base maintains vertical loading
- Edge clamp restraints to lower panel support
- Fixture does not require bonding to a plate
- Ability to test 1 in. to 3 in. wide sandwich specimens





Experimental Validation: Mode I Sandwich Panel Selection

		Facesheets					
		Unidirectional Prepreg			Woven		
_		3 Ply [0/90/0] _T	6 Ply [0/90/0] _{2T}	12 Ply [0/90/0] _{4T}	2 Ply [(0/90)/(±45)] _T	6 Ply [(0/90)/(±45)] _{3T}	10 Ply [(0/90)/(±45)] ₅₁
Cores	Nomex Honeycomb HRH10-1/8-8	х	х	х			
	Aluminum Honeycomb CR III-3/16-5052001	х	х	х			
	Balsa Wood Baltek S67					Х	
	Polyurethane Foam FR-6703						Х
	Polyurethane Foam FR-6710				Х	Х	

- Secondary bond with film adhesive used on honeycomb cores
- Single-step VARTM process used with foam and balsa cores



Mode I Results: Polyurethane Foam Core Sandwich

- Semi-stable delamination propagation
- No apparent effect of facesheet thickness on G_c





Mode I Results: Nomex Honeycomb Core Sandwich

Stable & Semi-stable delamination propagation
No apparent effect of facesheet thickness on G_c







Mode I Results: Aluminum Honeycomb Core Sandwich

- Stable & Semi-stable delamination propagation
- No apparent effect of facesheet thickness on G_c



Mode I Results: End-Grain Balsa Wood Core Sandwich

- Stable and Semi-stable delamination growth
- Growth arrested at breaks in interfaces between balsa core "blocks" (followed by unstable growth)



Mode I Single Cantilever Beam (SCB) Test: Specimen Width Effects

- Testing using three specimen widths
 - 1 in. 2 in. 3 in.
- Three core materials investigated
 - Aluminum honeycomb
 - Nomex honeycomb
 - Polyurethane foam





Crack front, polyurethane foam core





Mode I Single Cantilever Beam (SCB) Test: Width Effects With Honeycomb Cores







Mode I Single Cantilever Beam (SCB) Test: Width Effects With Polyurethane Foam Core







Specimen Width Effects: Anticlastic Curvature Due To Bending

- Crack front lagging on the free edges due to anticlastic bending of facesheet
- Anticlastic curvature highly dependent on v₁₂ of facesheets









Interlaminar normal stress at top surface of core

Vertical displacement of delaminated facesheet



Symmetry BC

Current Status:

Mode I Single Cantilever Beam (SCB) Test

- Appears well-suited for common sandwich configurations
 - High percentage Mode I
 - Delamination propagation along facesheet/core interface
 - Stable or semi-stable crack growth
- Width effect present due to anticlastic curvature
- Completing parametric study to identify recommended specimen geometries
- Composing a draft ASTM standard





SELECTED MODE II CONFIGURATION: *Hinged* Cracked Sandwich Beam (CSB)

- Three-point flexure loading
- Additional support provided at delaminated facesheet to create crack opening
- Relatively high percentages (>80%) of Mode II energy release rate produced
- Appears to be a suitable Mode II test method



Original Mode II Test Fixture

- Modified three-point flexure fixture
- Roller used to prevent facesheet/core interaction
- Required removal of core to place roller







Further Test Fixture Development: Mode II Testing



- Spool assembly allows tensioning of wire
- Adjustable height of wire





Prototype Mode II Test Fixture: Hinged Cracked Sandwich Beam (CSB)







Mode II Test Results: Foam and Nomex Honeycomb Cores

Semi-stable delamination propagation



Mode II Test Results: Aluminum Honeycomb Core

Core failure in aluminum honeycomb prior to delamination growth



Facesheet Stiffness Effects: Mode Mixity Variations Across Specimen Width

- 3D finite element analysis of 4 in. wide specimen
- Two facesheet moduli values analyzed: 4.4 Msi and 11.6 Msi



Core Thickness Effects: Mode Mixity Variations Across Specimen Width

- 3D finite element analysis of 4 in. wide specimen
- Three core thicknesses investigated



Addressing Mode Mixity/Width Variation: Adding Flexural Stiffness to Bottom Facesheet

Increasing flexural stiffness (EI) of lower portion of delaminated specimen reduces specimen width effect



Increasing Flexural Stiffness to Bottom Facesheet: Asymmetric Tabbing of Sandwich Composite



Tabbing of bottom of sandwich with conventional tabbing material G10 glass-epoxy expected to produce acceptable flexural stiffness ratio





CURRENT FOCUS: Sensitivity Studies on Specimen Parameters

- Determination of Acceptable Ranges of Sandwich Configurations
 - Facesheet parameters
 - Thickness, flexural stiffness, flexural strength
 - Core parameters
 - Thickness, stiffness, strength
 - Specimen and delamination geometry
- Composing draft ASTM standards



