

Lightning strike damage to composite structures

PI: Paolo Feraboli

*Aeronautics & Astronautics
University of Washington*

Boeing focal: Stan Alton

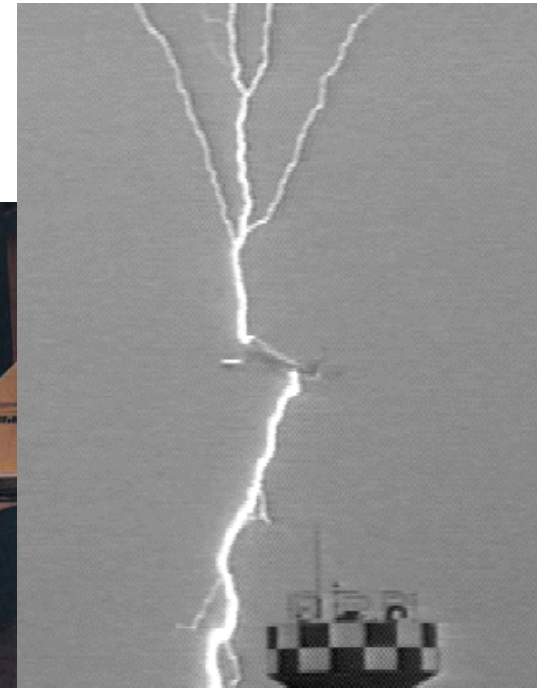
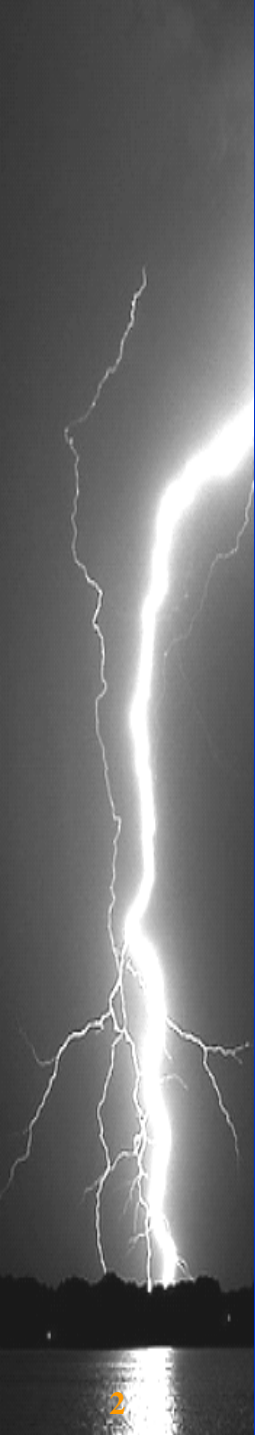
*Boeing Commercial Airplanes
787 Technology Integration*



*Presented at the AMTAS Spring Meeting
March 20th, 2008*

Outline

- *Importance of Lightning Strike research*
- *Current Lightning Strike facilities in the US*
- *Damage tolerance requirements*
- *Proposed research*
- *Status of research at UW*
- *Acknowledgments*

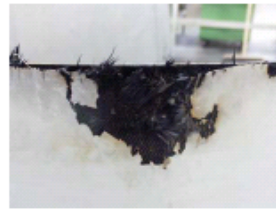


Importance of Lightning Strike research

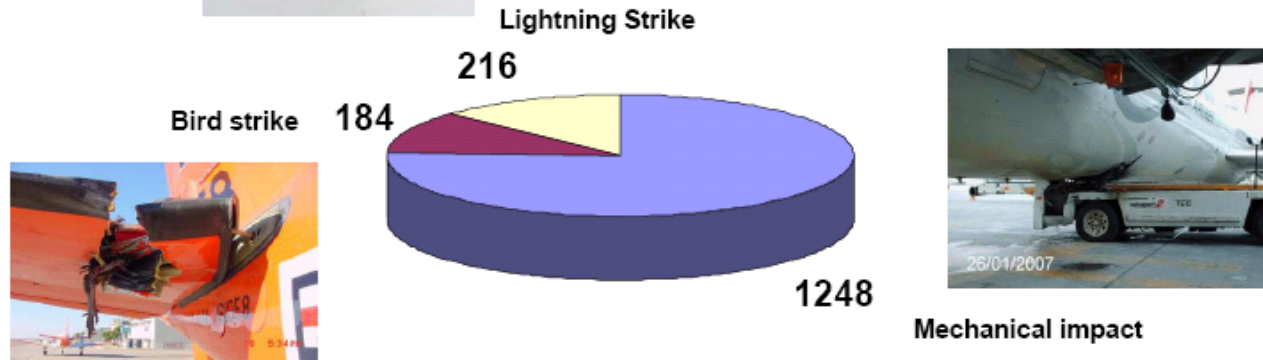
- *Carlos Blohm (Lufthansa Technik)*
 - *Presented at the FAA/ Boeing/ Airbus/ CACRC joint damage tolerance and maintenance meeting*

Damages on composite components

Total damages – Total fleet – Year 2006

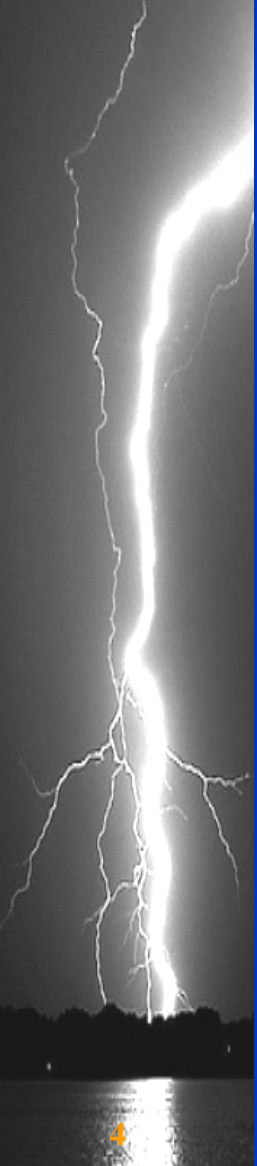


Fleet Size: 243 a/c
Total damages: 1647 events
Repair Cost: € 33.200.000



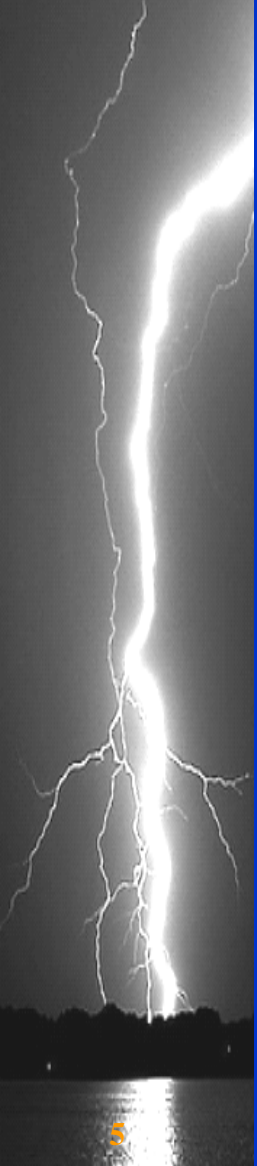
Importance of Lightning Strike research

- *Composite academic community has not addressed the issue*
 - *Non-existent literature in scholarly publications*
- *Extensive research in 1970's and 1980's*
 - *NASA, FAA and SANDIA*
- *Extensive research in proprietary domain*
 - *Boeing has active labs but highly sensitive nature of topic*
- *AGATE – delivered handbook in 2002*
 - *LTI performed contract research*
 - *Focused mostly on thin-gage sandwich*
 - *Large focus on glass fiber skins*



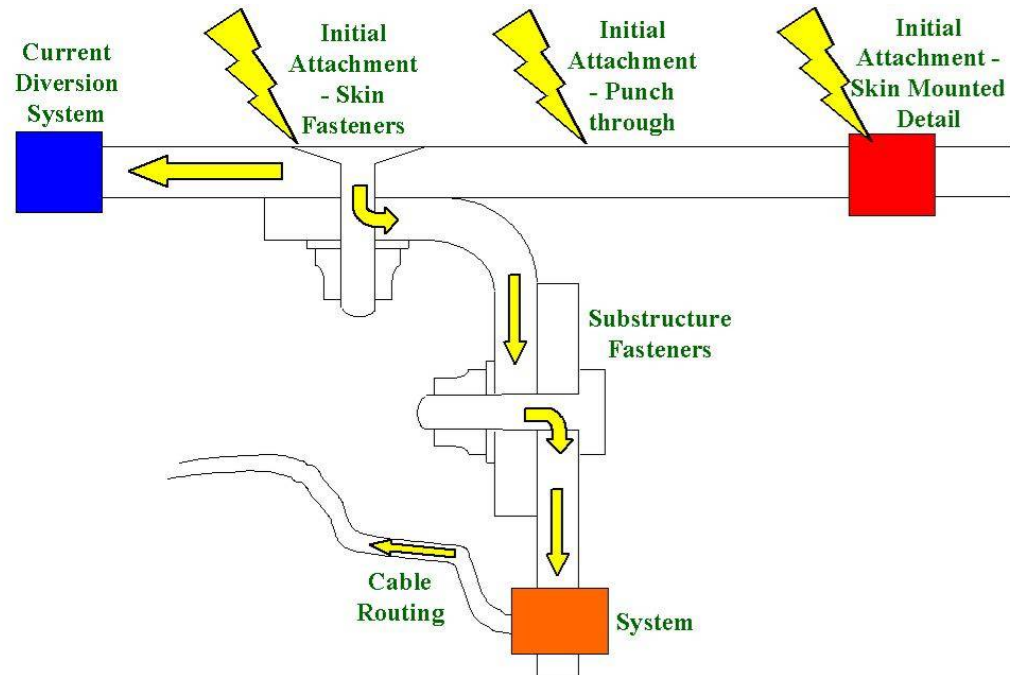
Current Lightning Strike facilities in the US

- *Full-size facilities*
 - *LTI – Pittsfield, MA*
 - *DNB – Fullerton, CA*
 - *Boeing Commercial – Seattle, WA*
 - *Shaw Aero Devices** – Naples, FL*
 - *Goodrich** – Chula Vista, CA*
 - *Red Stone/ Marshall** – Huntsville, AL*
- *Wichita State Univ. (NIAR)*
 - *Currently only indirect effects*
 - *Building facility for direct effects testing – online late 2008*
- *Caltech*
 - *Table-top generator 20kAmps for ignition studies*



A systems approach

- Indirect effects
 - EMI shielding
- Direct effects
 - Structural damage
 - Sparking in fuel tank
- Materials, fasteners, sealants, etc. all contribute



Spirit Aero (Wichita)

1/8 Korex core, S-2 glass skin

With Al mesh

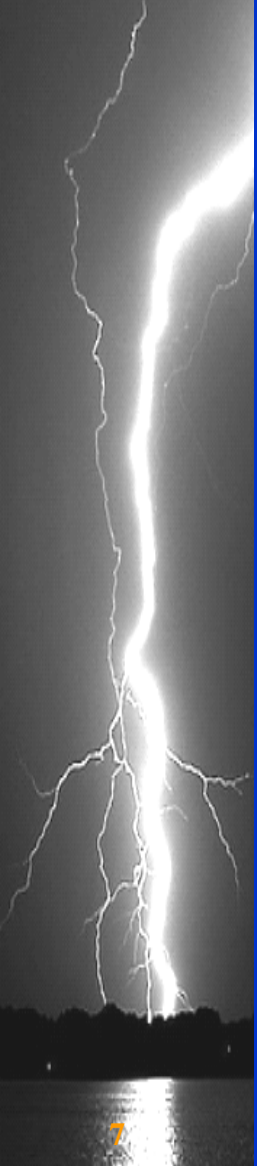


Before Zone 1A strike



After Zone 1A strike – LSP-001

Without Al mesh



Spirit Aero (Wichita)

1/8 Korex core, S-2 glass skin

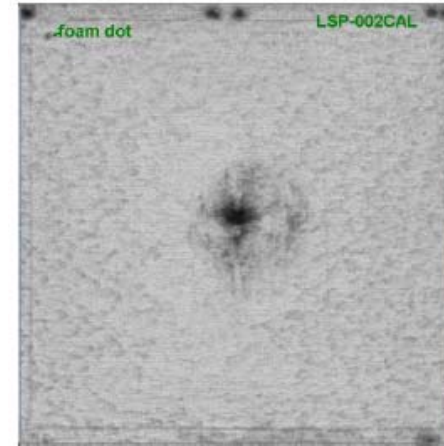


With Al mesh: 3-4 in. damage size,
limited to outside mesh ply and paint

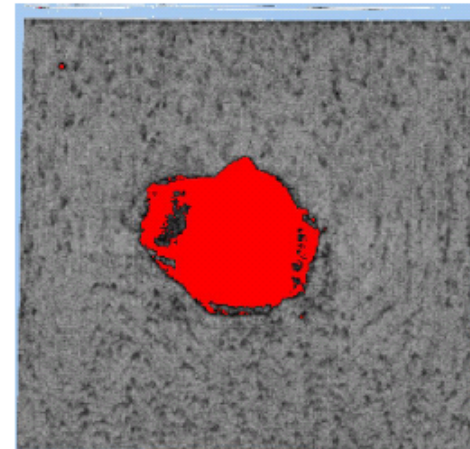


Without Al mesh: 9 in. damage size,
and punctured both skins

1-2 inch damage, mesh only
-validated via TTU (NDI)

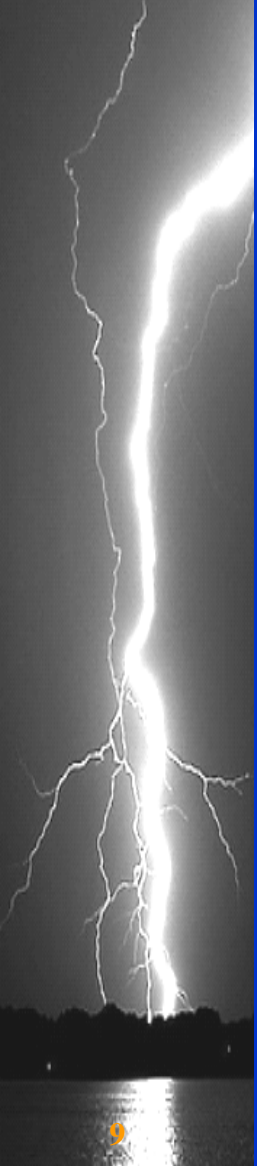


Damage area 12 inches, plus thru hole –
Validated by TTU



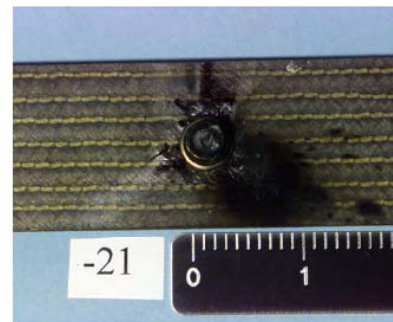
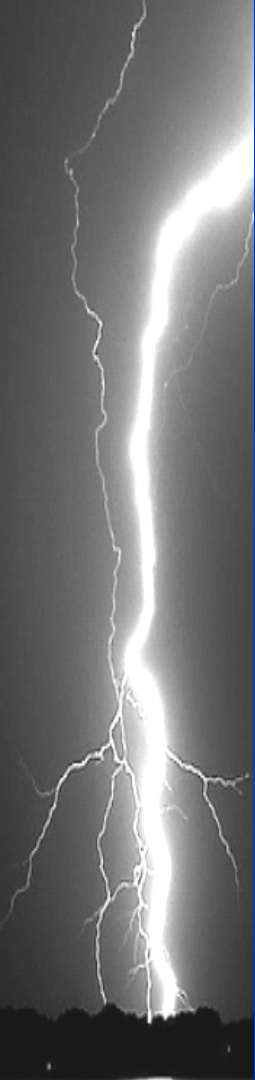
Proposed research

- *Current LS research almost exclusive domain of EE engineers*
- *Focus on:*
 - *Indirect effects*
 - *Sparking and fuel ignition*
 - *Systems/ certification of assemblies and components*
 - *Structures-focused work has been more superficial*



Proposed research

- *PI proposes to focus on structural aspects ONLY*
 - *Laminate carbon fiber laminates*
 - *Damage resistance (traditional and novel protection systems)*
 - *Damage tolerance (OHT/ OHC, and other residual strength)*
 - *Joint detail (effect of fastener preload, effect of fit, etc.)*
 - *Moisture and temperature effects*
 - *Different material forms (high vs. standard modulus fibers, toughening interlayer, cure temperature, stacking sequence issues)*
 - *Repair issues*



Proposed research

- *Medium size 100 kAmps, 40 kV generator that can test small test articles (up to 12 in x 12 in)*
- *Building block: certify by analysis supported by test evidence*
 - *Improve the foundations of the pyramid (basic level) in order to facilitate transition to different material forms*
 - *Standardize analysis tools to streamline certification process*

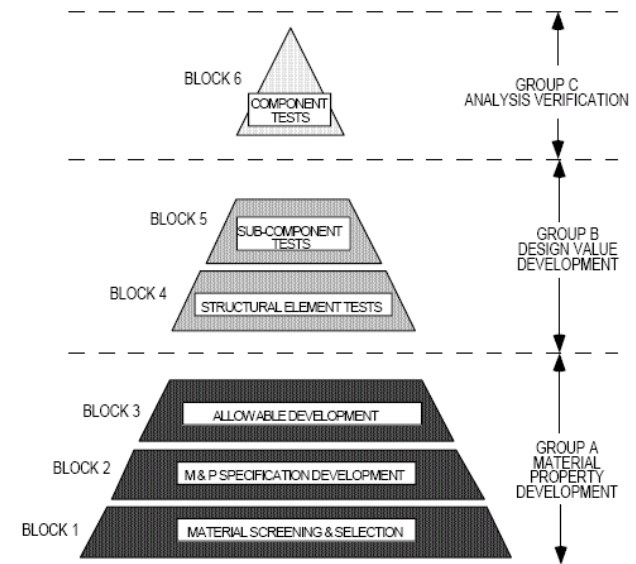
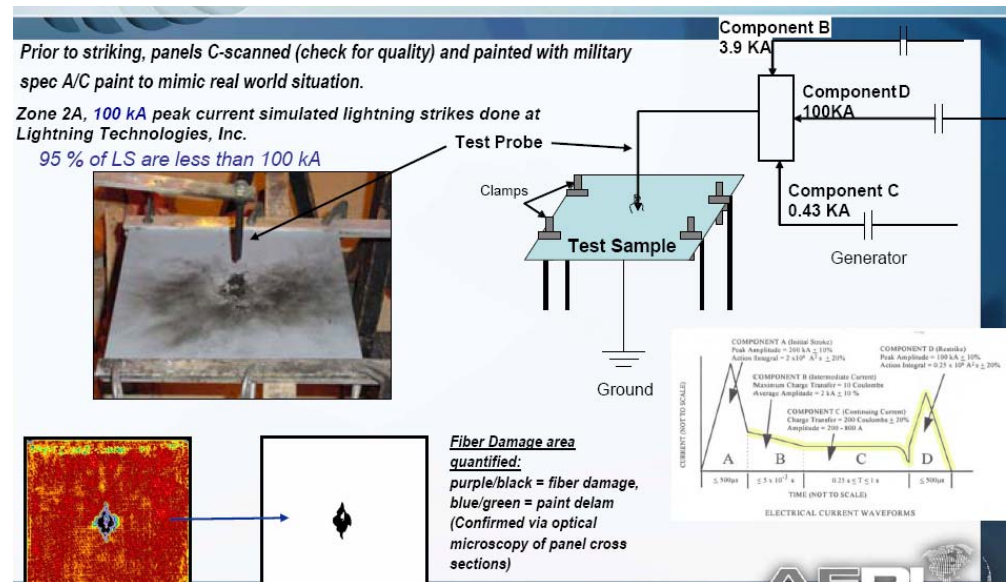


FIGURE 4.4.3.2 Building block approach.

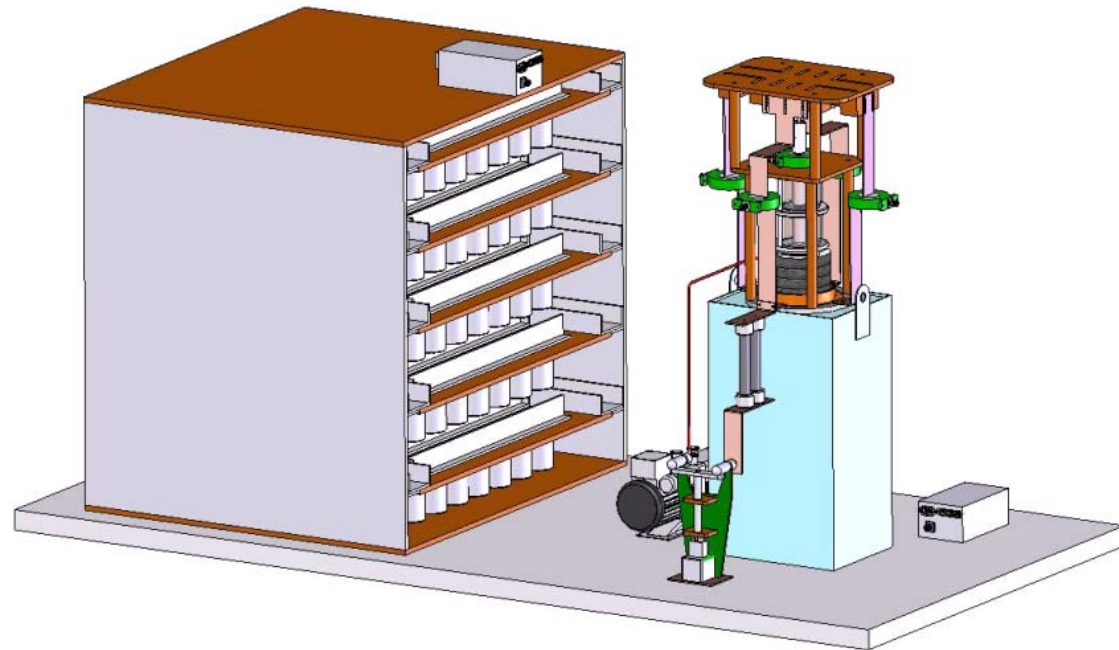
Inter-agency collaboration

- **Air Force:**
 - Les Lee of AFOSR and Tia Benson-Tolle of AFRL/WPAFB are currently involved in lightning activities
 - They use LTI test house facilities
 - Emphasis in multifunctional materials and structures, and nano-composite additives for improved conductivity
- Recent CMH-17 activities suggest expansion of participation of AFRL to safety and certification activities



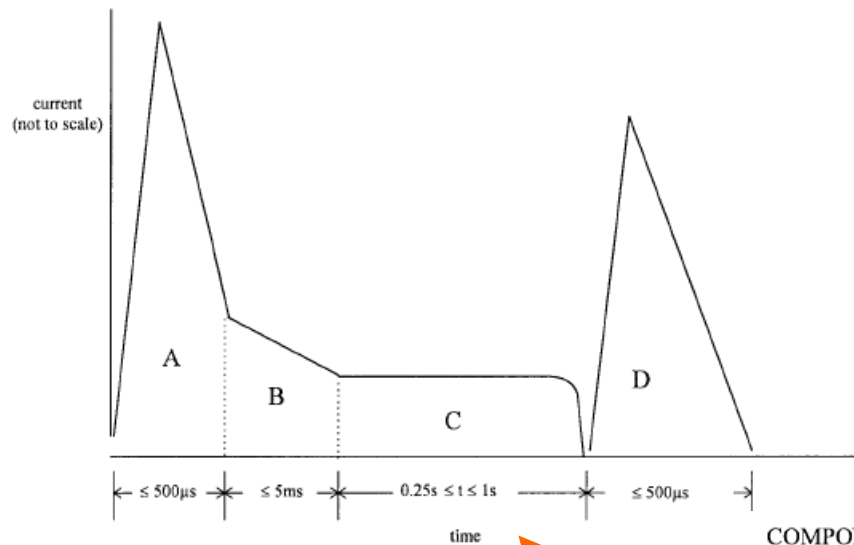
Status of research at UW

- *LS generator being built*
- *Boeing EME lab has provided guidance on lay-out and safety of circuit*



Status of research at UW

- *Waveform D demonstrated*
- *Waveform C will be online by April 2008*



COMPONENT A (First Return Stroke)

Peak Amplitude	:	200kA ($\pm 10\%$)
Action Integral	:	$2 \times 10^6 \text{A}^2\text{s}$ ($\pm 20\%$) (in $500\mu\text{s}$)
Time Duration	:	$\le 500\mu\text{s}$

COMPONENT B (Intermediate Current)

Max. Charge Transfer	:	10 Coulombs ($\pm 10\%$)
Average Amplitude	:	2kA ($\pm 20\%$)
Time Duration	:	$\le 5\text{ms}$

COMPONENT C (Continuing Current)

Amplitude	:	200 - 800A
Charge Transfer	:	200 Coulombs ($\pm 20\%$)
Time Duration	:	0.25 to 1 s

COMPONENT D (Subsequent Return Stroke)

Peak Amplitude	:	100kA ($\pm 10\%$)
Action Integral	:	$0.25 \times 10^6 \text{A}^2\text{s}$ ($\pm 20\%$) (in $500\mu\text{s}$)
Time Duration	:	$\le 500\mu\text{s}$

Acknowledgments

- *The Boeing Co.*
 - *Dale Winter, Diane Heidlebaugh, Rob Steinle, and Art Day*
 - *Patrick Stickler, Al Miller*

- *UW principal researcher*
 - *Mark Miller, MS student*

- *UW support group*
 - *Dave Medendorp, Sr. student*
 - *Andrew Southworth, Sr. student*
 - *Robert Gordon, engineer*
 - *Art Blair, engineer*

