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First Year Project

- **The Effect of Surface Treatment on the Degradation of Composite Adhesives**
 - **Surface treatment**
 - **Strength**
 - **Fracture toughness**
 - **Durability**
 - **Accelerated test methods**
 - **Wedge crack**

Coupons

- **Double cantilever beam**
 - **Mode I fracture toughness**
- **Wide area lap shear**
 - **Static load (s)**
 - **Low frequency (0.01 hz) cyclic load**
- **Wedge crack coupons**
 - **Compliant adherends**

Materials

- **Low cost prepreg adherends**
 - **777 or 7e7**
- **Co-bonded adhesive films**
 - **Metal bond 1515-3 or AF555**
- **Surface treatments**
 - **No prep (6001 peel ply)**
 - **Grit blast**
 - **Plasma etch**

Simulated Environmental Exposure

- **140F water immersion**
- **Up to 10k hour durations**

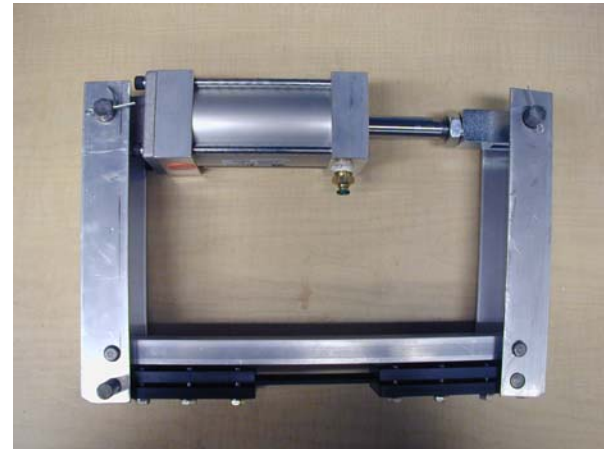
- **Post mortem surface characterization to be performed at University of Washington**

Environmental Exposure Facility

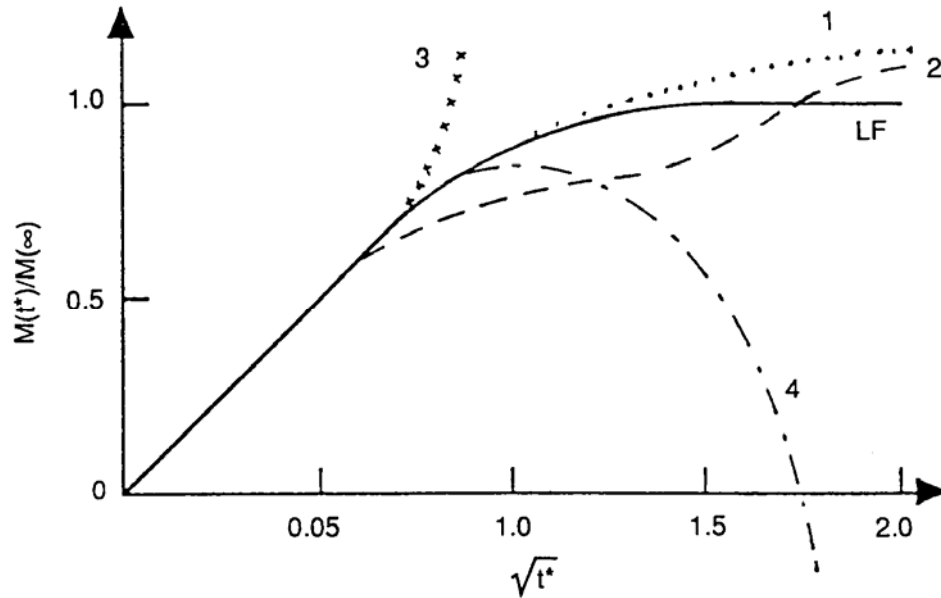
- Aggressive environments
- Temperature
- Stress
- Time



Environmental Exposure Facility



Fickian and Non-Fickian Weight Change



Modeling Degradation

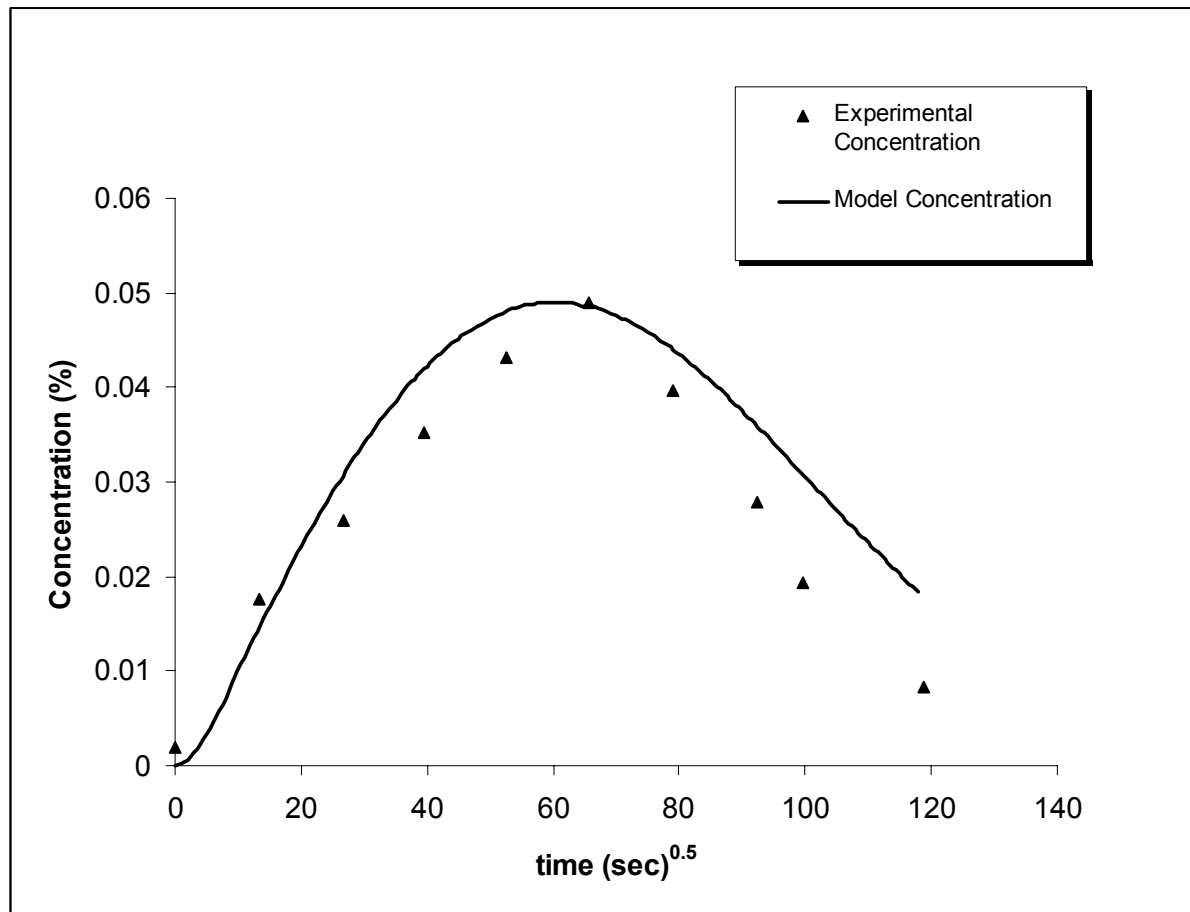
$$\frac{\partial c}{\partial t} = D \cdot \frac{\partial^2 c}{\partial z^2}$$

$$-\frac{d[C]}{dt} = k[C]$$

$$C(t, z) = \int_0^t \exp(-k(t - \tau)) \cdot \left[\frac{dC_o}{d\tau} \right] \cdot d\tau$$

$$\bar{C}(t) = \frac{1}{h} \int_0^h C(t, z) \cdot dz$$

Comparison of predicted and measured moisture concentration



Proposed First Year Project

- **Long term durability**
 - **Select a “young” composite material with multiple cure schedules**
 - **Load matrix (lap shear or +/- 45 degree coupons) at service temperature in aviation environments**
 - **Compare weight gain, strength and modulus retention as a function of time for disparate cure schedules**

Proposed First Year Project

- **Modeling degradation**
 - **Expose material to aggressive solvent with measurable weight change (gain or loss)**
 - **Characterize fundamental degradation parameters (D , c_m , k)**
 - **Consider temperature and geometry effects**